CEE 4674 Airport Planning and Design

Geometric Design: Part 2 Runway Exit Design and Exit Locations

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Geometric Design of Runway Exits

Sources of information:

- FAA AC 150/5300-13B (Chapter 4)
 - Sections 4.8.2 through 4.8.6
- ICAO Aerodrome Manual Volumes 1 and 2

Design principles:

- Provide ample space for aircraft to maneuver out of the runway
- Locate runway exits consistent with aircraft landing performance

Important Issues about Runway Exits

- Runway exits are responsible for making operations more efficient on the ground
- Poorly designed runway exits add valuable service time (i.e., runway occupancy time)
- Poorly placed runway exits can contribute to go-arounds and runway incursions
- Runway occupancy time and its standard deviation are critical parameters for runway capacity estimation

Runway Occupancy Time

- The time elapsed between the aircraft crossing the runway threshold and the time when the same aircraft crosses the imaginary plane of a runway exit paved area (aircraft wingtip or tailtip)
- Issues about ROT
 - The definition of ROT has been used inconsistently throughout the years
 - Many early ROT studies failed to recognize that when an aircraft starts turning towards the runway exit, the aircraft is still using the runway until its wingtip clears the runway edge plane

Factors Affecting Runway Occupancy Time

- Aircraft type (AAC group)
- Runway geometric design factors
 - Runway width
 - Pavement condition (wet, dry, contaminated)
- Taxiway geometry design factors
 - Number of runway exits on the runway
 - Taxiway type
 - Taxiway network interaction
- Pilot technique
- Traffic pressure (i.e., having another aircraft on short final behind)
- Gate location

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Source of Runway Occupancy Time Data FAA and Virginia Tech Landing Event Database





Landing Events Database

- Landing Events Database archives 36 million landing records from ASDE-X data (all landings at 43 U.S. airports during for years 2015-2020)
 - Stand-alone product (client software)

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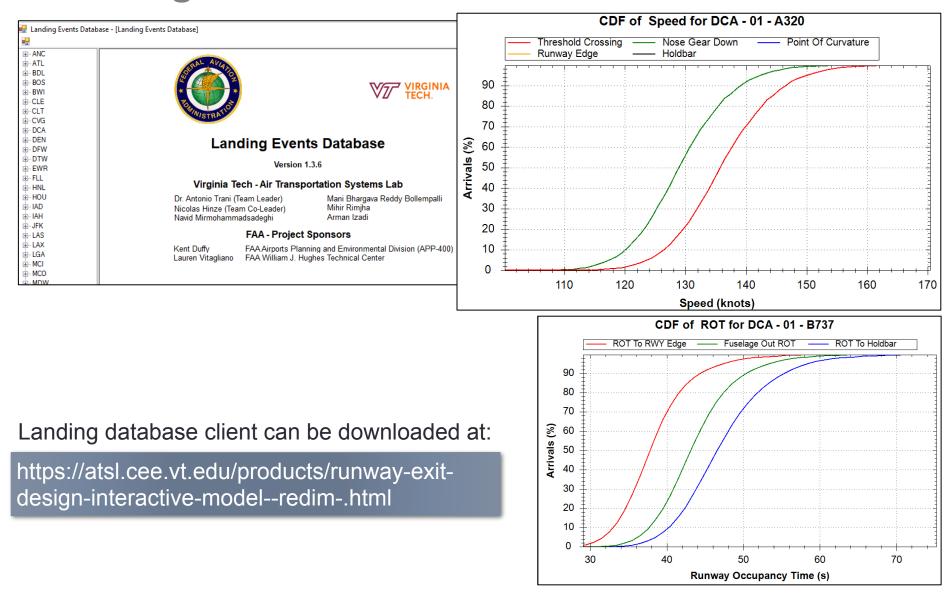
 Tabular and graphical data on runway exit utilization at 43 U.S. airports

Landing database client can be downloaded at:

https://atsl.cee.vt.edu/products/runway-exitdesign-interactive-model--redim-.html

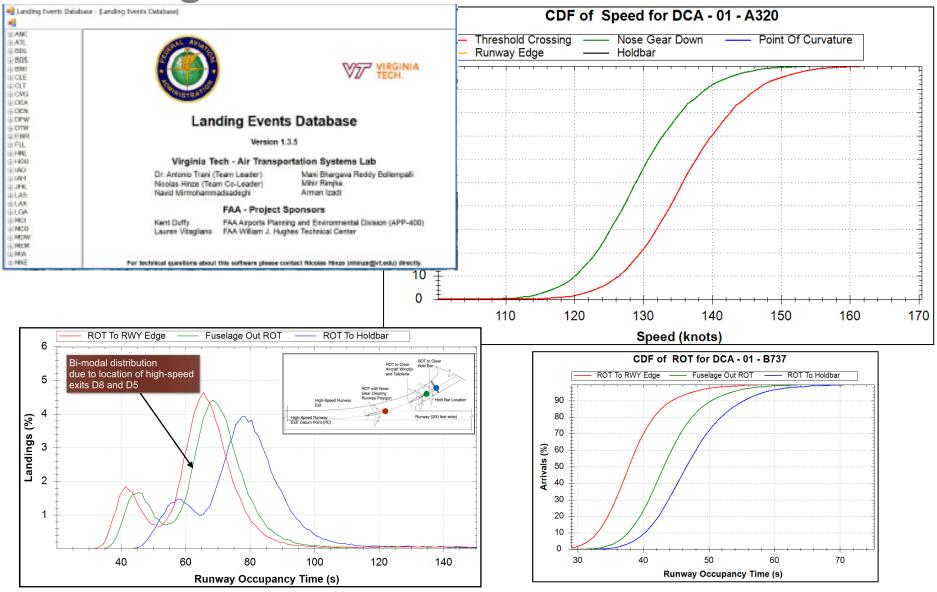


Landing Event Database Tool Versions 1.3.6-1.3.7



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Landing Event Database Tool : Screen Shots



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Landing Event Database Tool (1)

Analysis	Purpose	Metrics and Ready-Made Query Options	
Aircraft Mix	Provides an overview of aircraft fleet mix in the form of a pie chart with the top 10 aircraft in the fleet mix presented.	By runway By runway exit	
Runway Occupancy Time	Provides three values of runway occupancy time measured at three locations: 1.Runway edge 2.Fuselage out 3.At hold bar	 1.Average ROT (in seconds) by runway, runway exit and aircraft 2.Median ROT (in seconds) by runway, runway exit and aircraft 3.Probability Density Function (PDF) of ROT (dim) by runway, runway exit and aircraft 4.Cumulative density function of ROT by runway, runway exit and aircraft 5.Runway exit utilization (percentage) by runway exit and aircraft 	
Speed	 Provides information about five aircraft ground speeds at different locations of the landing profile: 1.Threshold 2.Nose gear down 3.Point of curvature 4.Runway edge 5.Hold bar 	 1.Average ROT (in seconds) by runway, runway exit and aircraft 2.Median ROT (in seconds) by runway, runway exit and aircraft 3.Probability Density Function (PDF) of ROT (dim) by runway, runway exit and aircraft 4.Cumulative density function of ROT by runway, runway exit and aircraft 5.Detailed speed profiles as a function of distance by aircraft, runway and runway exit 6.Detailed speed profiles as a function of time by aircraft, runway and runway exit 	
Nose Gear Location	Provides estimates of nose gear distance. The nose gear distance is estimated in the landing algorithm to initiate the nominal deceleration.	 1.Nose gear distance from runway landing threshold by runway, aircraft and runway exit 2.Probability Density Function (PDF) of nose gear distance (feet or meters) by runway, runway exit and aircraft 3.Cumulative density function of nose gear distance (feet or meters) by runway, runway exit and aircraft 	



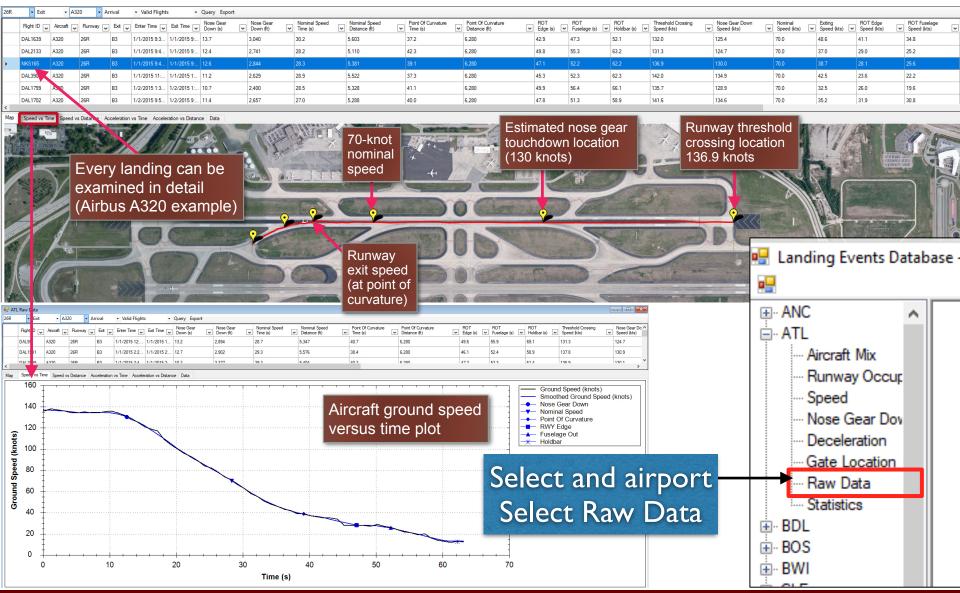


Landing Event Database Tool (2)

Analysis	Purpose	Metrics and Ready-Made Query Options	
Deceleration	Provides two values of aircraft deceleration on	Average deceleration (in m/s2) by runway, runway exit and aircraft	
	the runway:	Median deceleration (in m/s2) by runway, runway exit and aircraft	
	Nominal	Probability Density Function (PDF) of deceleration (in m/s2) by	
	Nominal location to point of curvature (Nominal to PC)	runway, runway exit and aircraft (both average and median values can be plotted)	
		Cumulative density function of aircraft deceleration (in m/s2) by runway, runway exit and aircraft (both average and median values can be plotted)	
Raw Data	Provides detailed information (in a table) on 30 key parameters for every landing contained in the Landing Events Database.	30 key parameters defining the landing profile of each landing operation. Parameters include: flight ID, aircraft type, runway, runway exit use, time of operation, nose gear touchdown distance	
	Provides graphical information of every landing in the database.	and time, nominal deceleration, deceleration from nominal point to PC, exit speed, and airport wind conditions.	
	Provides a graphical depiction of individual	Speed-distance profile of each landing event	
	landings in a Microsoft NAVTEQ map layer	Speed-time profile of each landing event	
	(bottom viewport)	Acceleration-time profile of each landing event	
		Acceleration-distance profile of each landing event	
		Processed numerical data with speed, acceleration, distance and time for individual landings.	
Statistics	Summarizes the landing statistics processed by	Total landing records	
	airport by month.	Valid records	
		Number of records with missing parameters	
		Number of records with unreasonable parameters	
		Records with no associated runway	
		Go-around records	





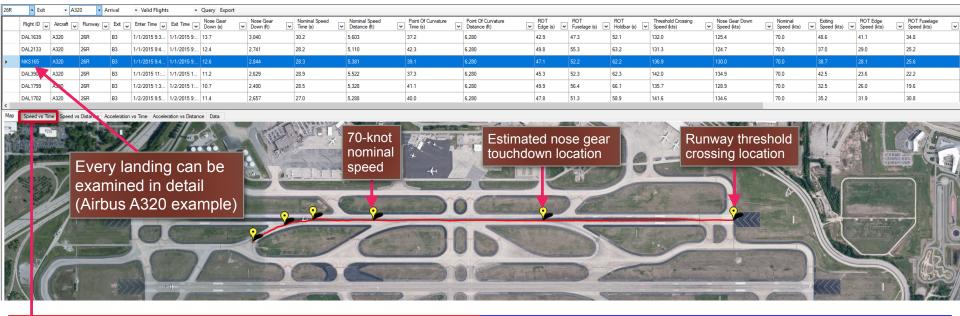


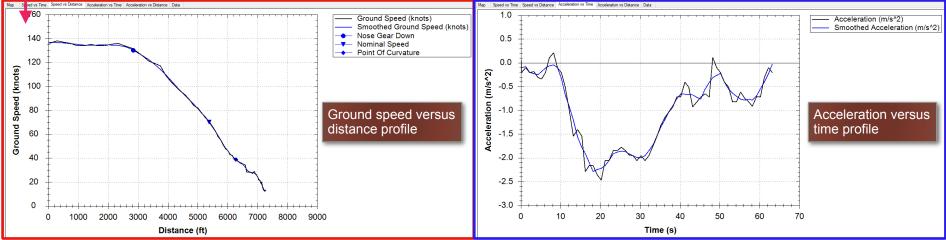
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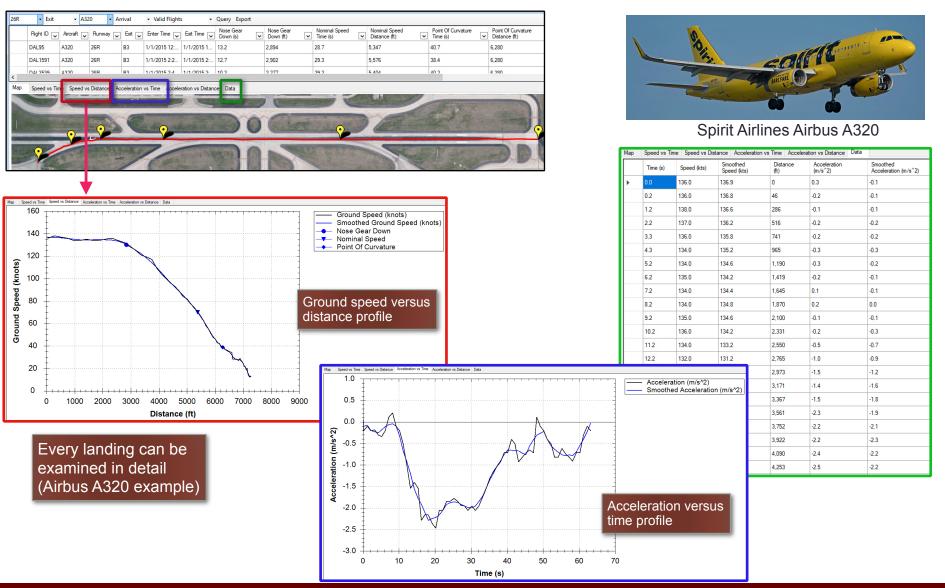
Landing Database Raw Data Viewer







Landing Database Raw Data Viewer (2)

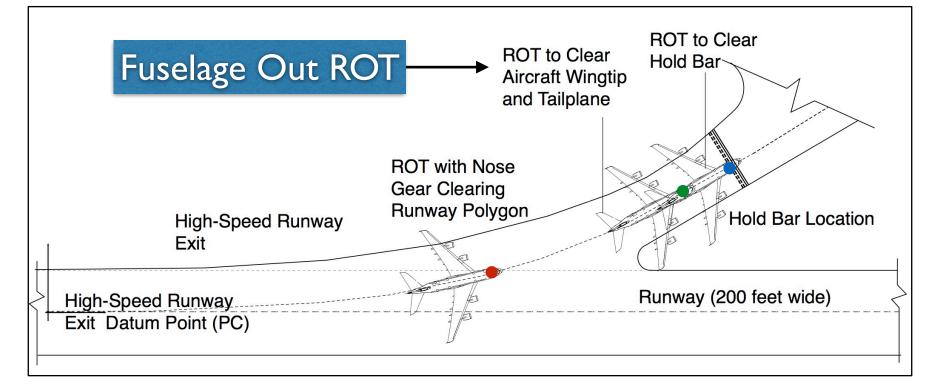


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Definitions of Runway Occupancy Time



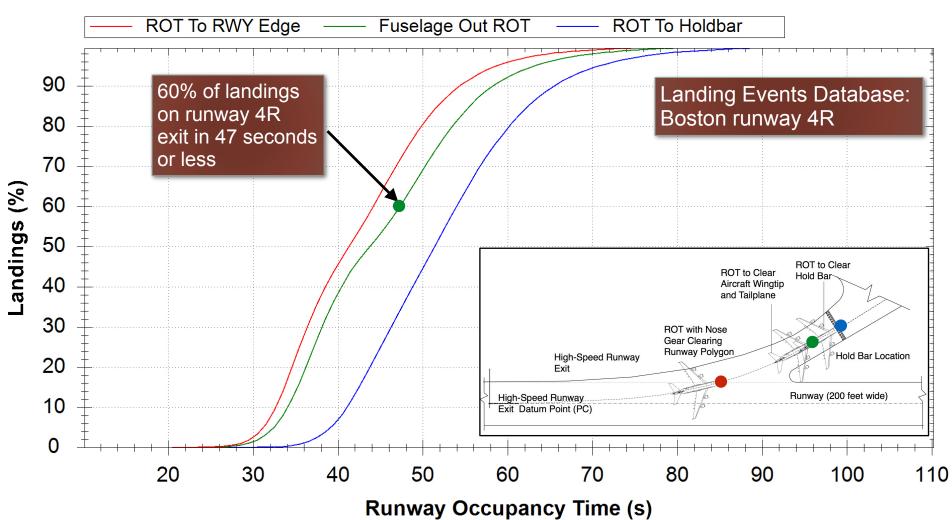
- In the US the definition of ROT is usually the time to vacate the runway with fuselage out
- In Europe and other countries ROT is interpreted as the time to reach the hold bar



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Invent the Future

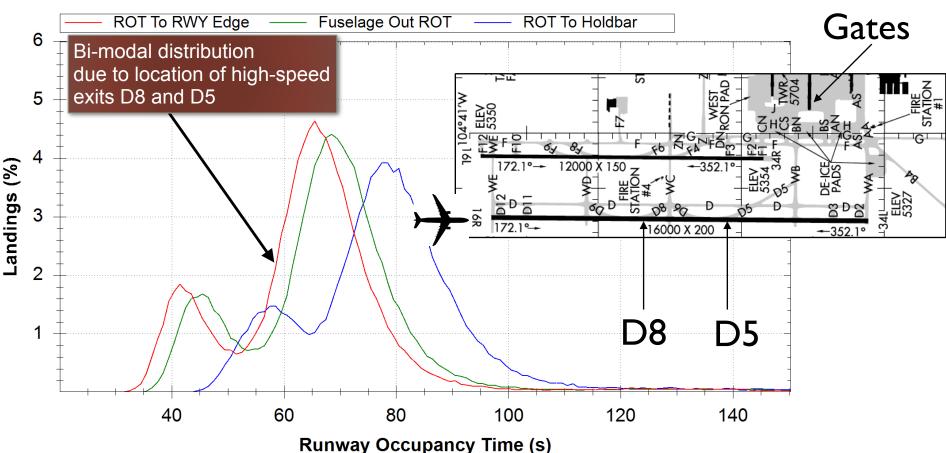
CDF of ROT for BOS - 04R



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Distribution of Runway Occupancy Times PDF of ROT for DEN - 16R

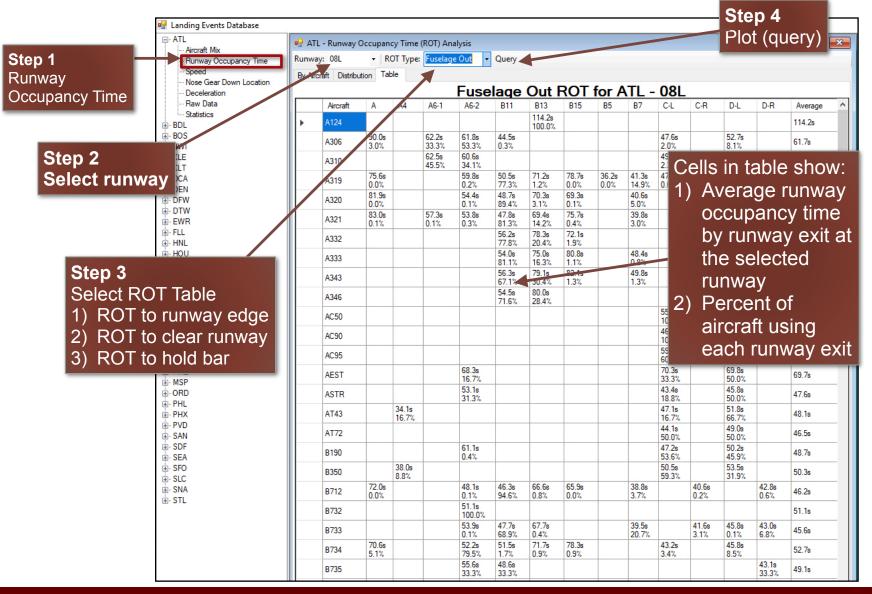
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Pilots are motivated to use high-speed runway exit **D5** because it reduces the taxiing time to the gate VirginiaTech



Runway Occupancy Time Tables

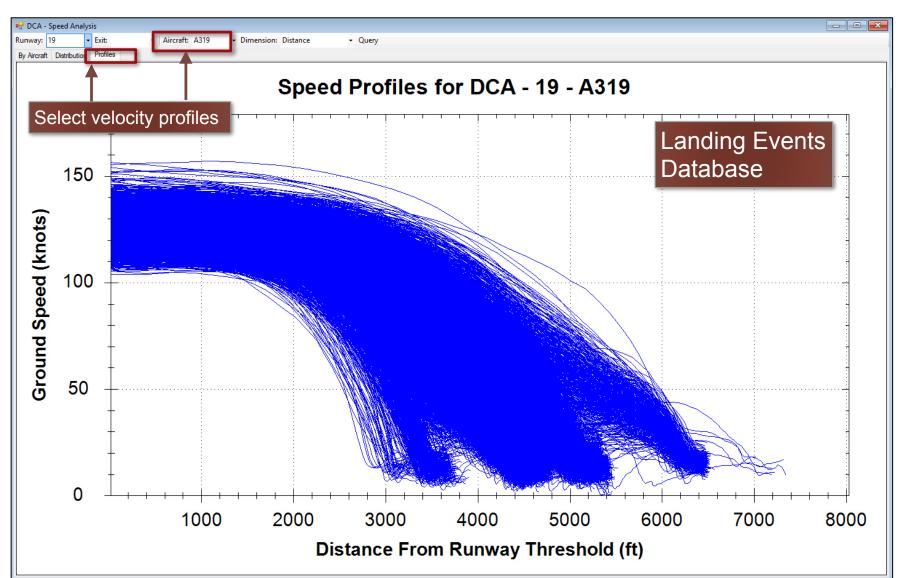


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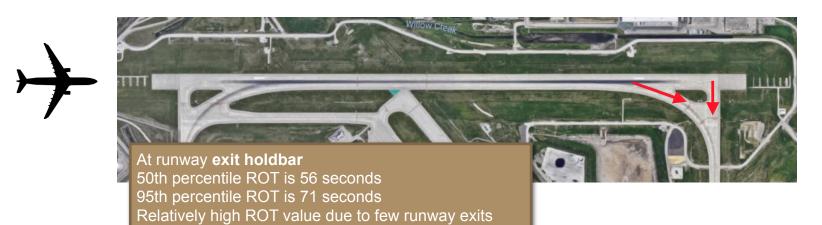
Aircraft Velocity Profiles : Airbus A319 at DCA Runway 19

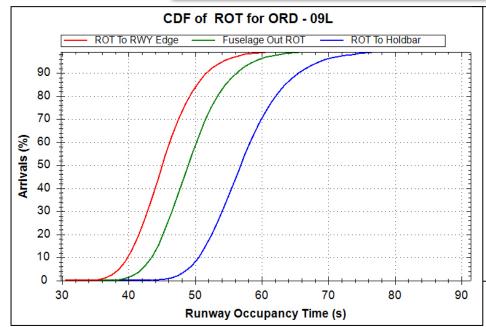


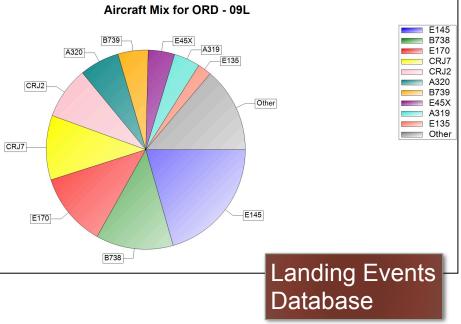




ORD Airport Runway 9L (Two Usable Exits)



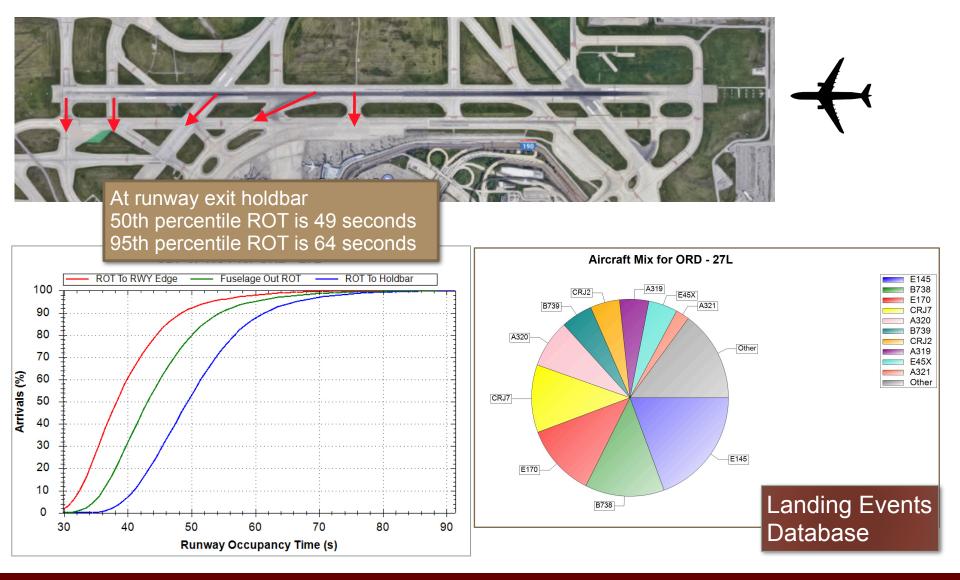


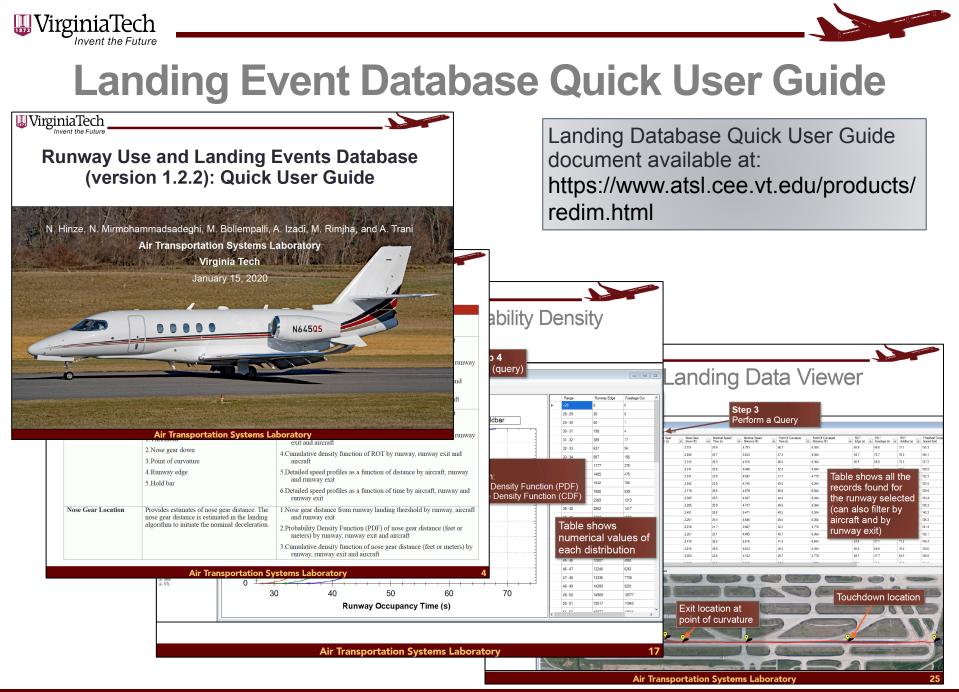






ORD Airport Runway 27L (Five Usable Exits)





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Runway Exit Design

Geometric Design Standards for Runway Exits Sources:

- FAA AC 5300-13B (Chapter 4)
 - Sections 4.8.2 through 4.8.6
- ICAO Aerodrome Manual Volumes 1 and 2 Design principle:
- Provide ample space for aircraft to maneuver out of the runway
- Make the runway exits easily identifiable and usable

What is the Issue with Runway Exits?

- Runway exits are responsible for making operations more efficient on the ground
- Poorly designed runway exits add valuable service time (i.e., runway occupancy time)
- Poorly placed runway exits can contribute to goarounds and runway incursions
- Runway occupancy time and its standard deviation are critical parameters for runway capacity estimation

Definitions

• Runway Occupancy Time (ROT)

—The time elapsed between an aircraft crossing the runway threshold and the time when the same aircraft crosses the imaginary plane of a runway exit paved area

Issues about ROT

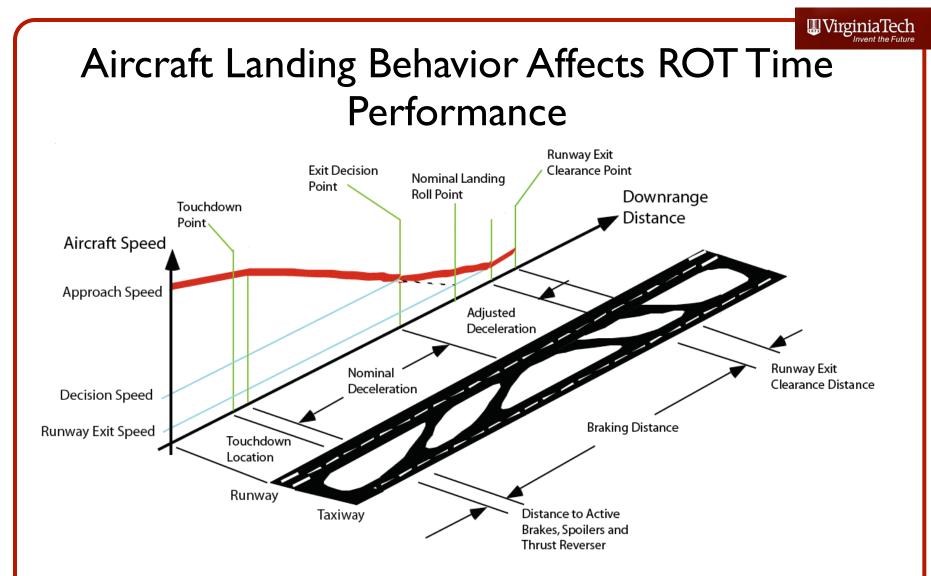
—The definition of ROT has been used inconsistently throughout the years

—Many early ROT studies failed to recognize that when an aircraft starts turning towards the runway exit, the aircraft is still using the runway until its wingtip clears the runway edge plane

Factors Affecting ROT

- Aircraft mix
 - Percent of aircraft in various runway performance groups
- Runway and runway exit geometric design factors
 - Runway width
 - Pavement condition (wet, dry, contaminated)
- Taxiway geometry design factors
 - Number of runway exits within the aircraft mix acceptability requirements
 - Taxiway type
 - Taxiway network interaction
- Pilot technique
 - Traffic pressure (i.e., having another aircraft on short final behind)
 - Gate location
 - Airport familiarity

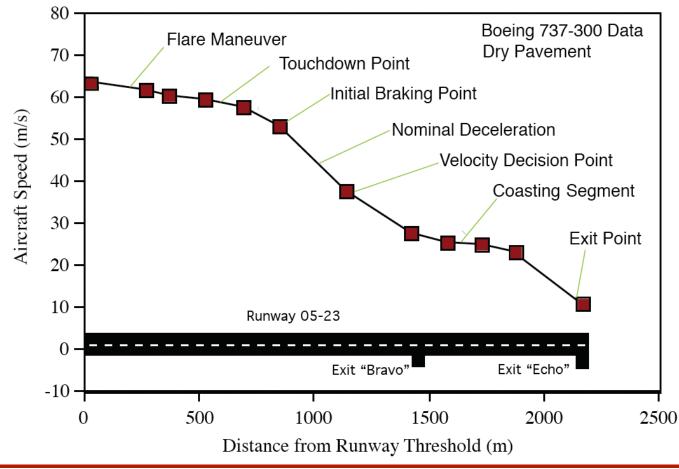
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For hand calculation analysis of runway exit placement, we can model the landing roll into three segments: air segment, free roll segment, and deceleration segment

Aircraft Landing Roll Profile

 Sample data collected at Charlotte-Douglas International Airport (CLT) Runway 05-23 (Trani et al., 1996)



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Typical Landing Milestones





Air phase starts Crosses the runway threshold Reference speed over threshold (Vref)

Air phase ends Main gear touchdown Typically 95% of threshold crossing speed

Deceleration phase starts Nose gear touchdown Engine thrusts reversers deployed Wing spoilers deployed Wheel braking on

Typical Landing Milestones (2)

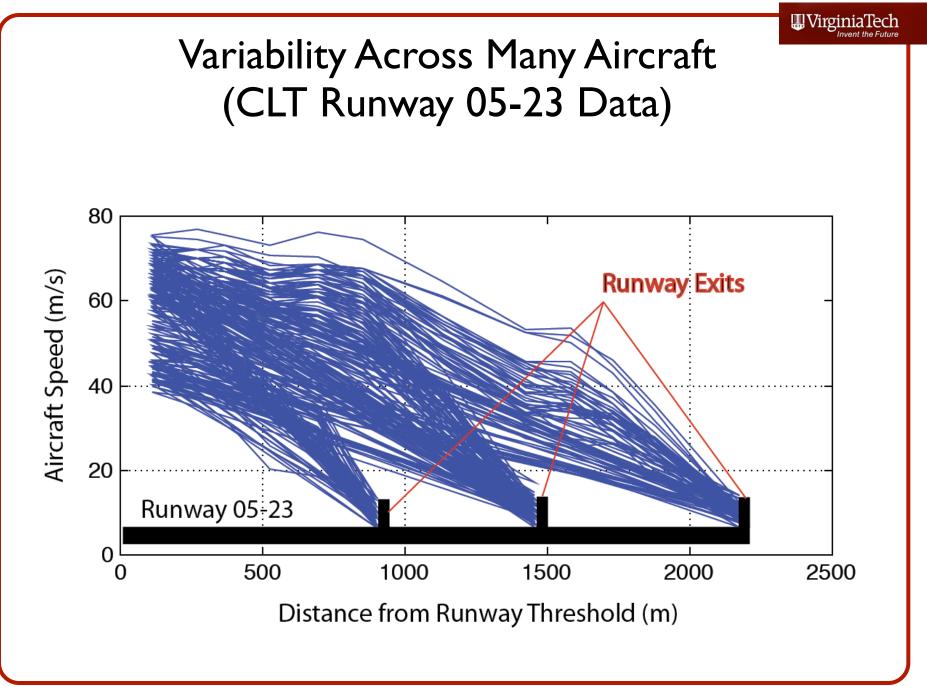


Landing deceleration segment Engine thrusts reversers deployed Wing spoilers deployed Wheel braking on



Turnoff maneuver Engine thrusts reversers stowed Wing spoilers retracted No wheel braking

Most jet powered aircraft manufacturers recommend thrust reversers to be stowed at 60 knots to avoid engine foreign object ingestion



Effects of ROT on Runway Capacity

- Modest gains in runway saturation capacity are possible with reductions in ROT because in today's environment, inter-arrival separations dominate over runway capacity
- ROT nevertheless is important in runways used with mixed operations (i.e., arrivals and departures) in both IMC and VMC conditions
 - Reduced weighted average ROT values reduce the gap needed to launch departures between successive arrivals
 - The same effect is true if reductions in the standard deviation of ROT are possible
- ROT is more important under VMC operations because inter-arrival times (IAT) are smaller compared to those observed during IMC conditions
- Standard deviation of ROT is very important
- Some small gains under IMC conditions (mixed operations in a single runway)

Runway Exits

- The purpose of runway exists is to improve service
- times of airport runways
- The number of runway exists varies from airport to
- airport and within runways at the same airport
- Several types of runway exits can be

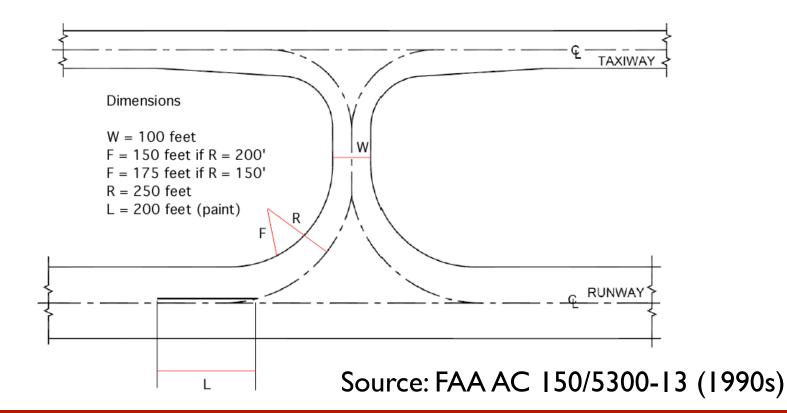
Types of Runway Exits

Runway Exit Type	Characteristics and Use	Remarks and Exit Speeds
Right-angle	Low volume of traffic	Low speed
(90 degree)	Ends of a runway	(5-8 m/s)
45 degree General Aviation	Old design (not recommended)	Medium speeds (8-15 m/s)
30-degree Constant	Two versions	High-speed
Radius Design	Use when > 30 operations/hr	15-23 m/s
30-degree Spiral Design	Adopted in the mid 80s	Transition spiral
	Use when > 30 operations/hr	15-23 m/s

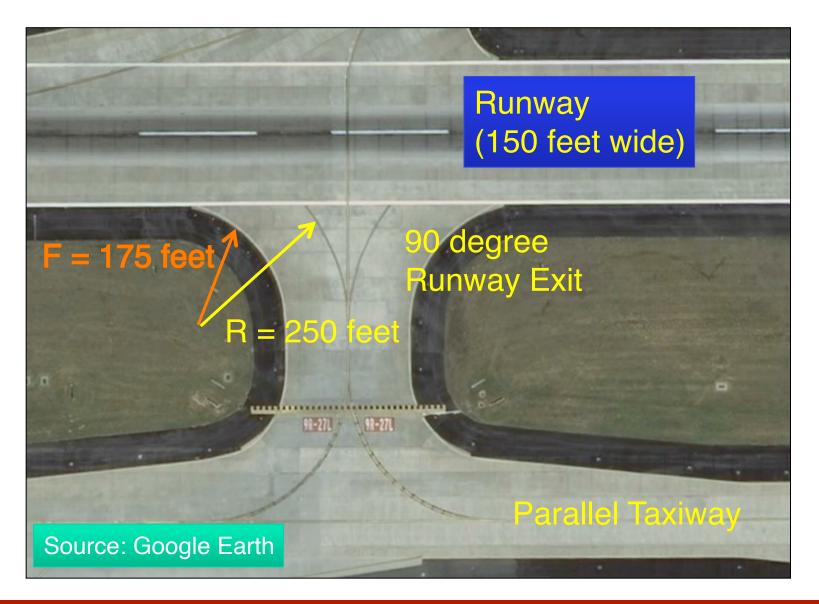
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Right-Angle Exits

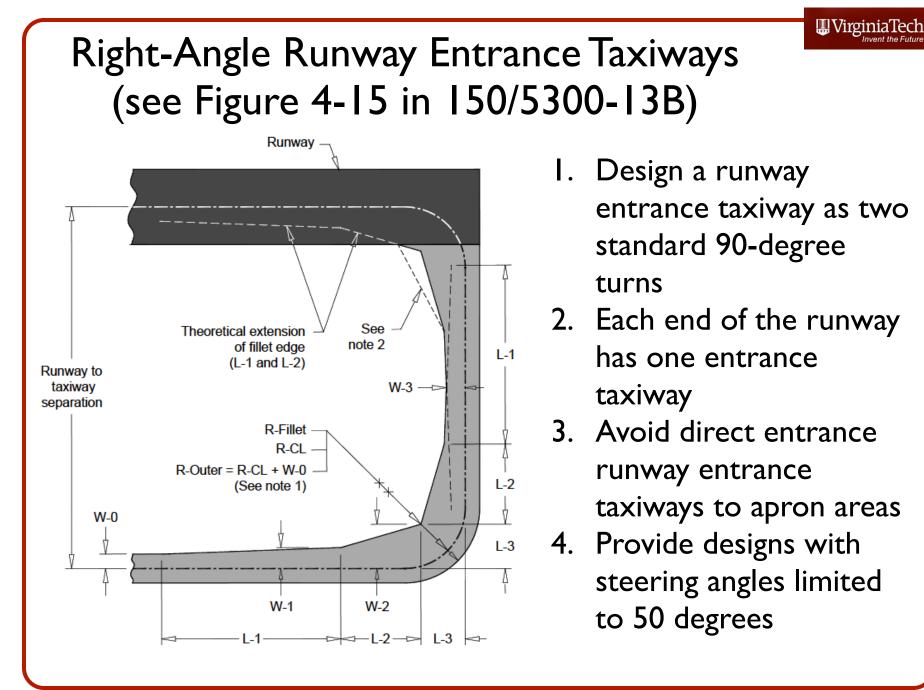
- Old (legacy) design baseline centerline radius was 250 feet
- Pavement edge radius varies according to runway width
- Legacy design has been replaced by new design practice



Sample Implementation (ATL)



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Right-Angle Runway Entrance Taxiway Dimensions when steering angles > 50

Dimension	TDG														
(see <u>Figure</u> <u>4-15</u>)		3			4				5			б			
Runway Centerline to Taxiway Centerline Distance (ft)	300	350	400	300	350	400	450	500	400	450	500	400	450	500	550
W-0 (ft)	25	25	25	25	25	25	25	25	37.5	37.5	37.5	37.5	37.5	37.5	37.5
W-1 (ft)	33	33	33	37	37	36	36	36	49	48	48	51	51	51	50
W-2 (ft)	55	54	54	80	77	75	74	74	84	83	83	102	100	98	97
W-3 (ft)	31	29	28	49	42	38	35	33	50	47	45	63	57	53	50
L-1 (ft)	173	172	172	315	310	307	305	305	301	299	298	408	405	403	401
L-2 (ft)	82	83	83	140	140	140	140	140	137	136	138	180	180	180	180
L-3 (ft)	55	54	54	80	77	75	74	74	84	83	83	102	100	98	97
R-Fillet	0	0	0	0	0	0	0	0	0	0	0	0	0	0	0
R-CL (ft)	62	61	60	110	102	98	95	94	98	95	94	133	126	122	120
R-Outer	87	86	85	135	127	123	120	119	135.5	132.5	131.5	170.5	163.5	159.5	157.5

Increase the centerline radius 2. Increase the fillet dimensions 3. use Tables 4-3 and 4-4 (labeled as **Dimensions** for Runway Entrance/Exit Taxiways with Two 90-Degree Turns are Nonstandard)

Runway Entrance Taxiways (TDG Groups 3-6)

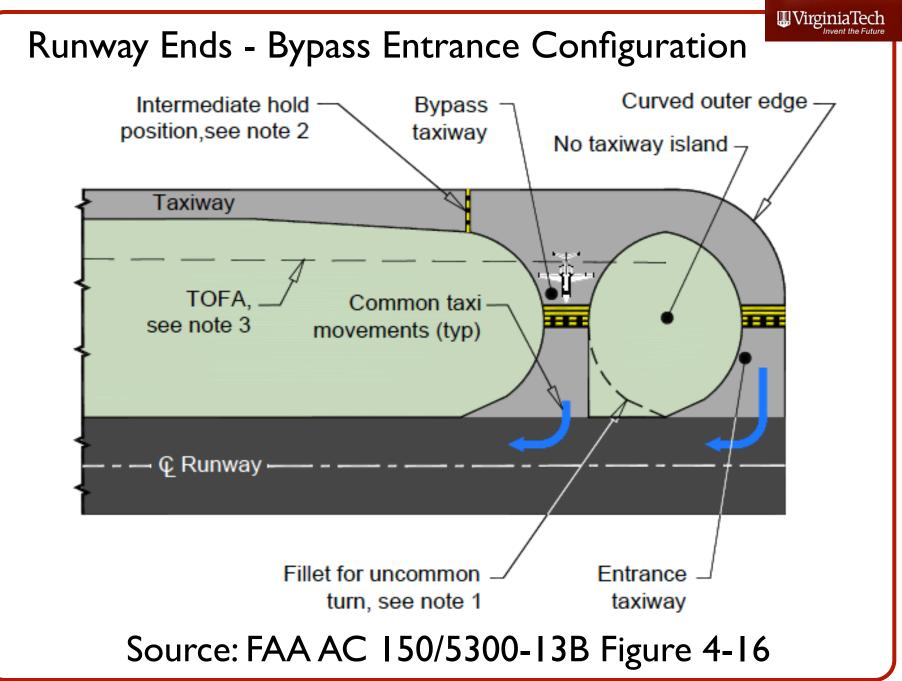
Dimension	TDG														
(see <u>Figure</u> <u>4-15</u>)		3			4				5			6			
Runway Centerline to Taxiway Centerline Distance (ft)	300	350	400	300	350	400	450	500	400	450	500	400	450	500	550
W-0 (ft)	25	25	25	25	25	25	25	25	37.5	37.5	37.5	37.5	37.5	37.5	37.5
W-1 (ft)	33	33	33	37	37	36	36	36	49	48	48	51	51	51	50
W-2 (ft)	55	54	54	80	77	75	74	74	84	83	83	102	100	98	97
W-3 (ft)	31	29	28	49	42	38	35	33	50	47	45	63	57	53	50
L-1 (ft)	173	172	172	315	310	307	305	305	301	299	298	408	405	403	401
L-2 (ft)	82	83	83	140	140	140	140	140	137	136	138	180	180	180	180
L-3 (ft)	55	54	54	80	77	75	74	74	84	83	83	102	100	98	97
R-Fillet	0	0	0	0	0	0	0	0	0	0	0	0	0	0	0
R-CL (ft)	62	61	60	110	102	98	95	94	98	95	94	133	126	122	120
R-Outer	87	86	85						135.5						157.5

Note: Use two standard 90-degree turns for combinations of TDG and common runway to taxiway separation, not shown in this table.

Source: FAA AC 150/5300-13B Table 4-4

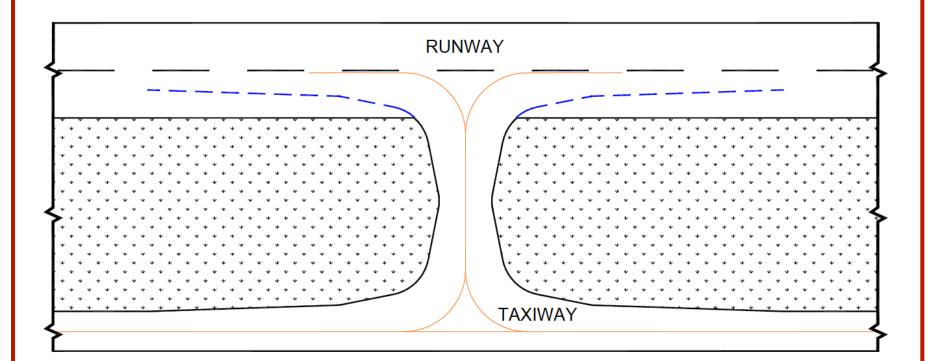
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Right Angle Runway Exit (see Figure 4-19)



New standard for right angle exits Build the exit as a projection of two runway entrance exits



Example Problem

- Design a new right-angle runway exit for a 8,000 foot runway (150 foot wide)
- Critical aircraft design is the Airbus A340-600
- For this airport the runway to taxiway centerline is 500 feet



Procedure

Step I - Identify the Airbus A340-600 as TDG-6

CMG = 121.6 feet MGW = 41.37 feet

Step 2

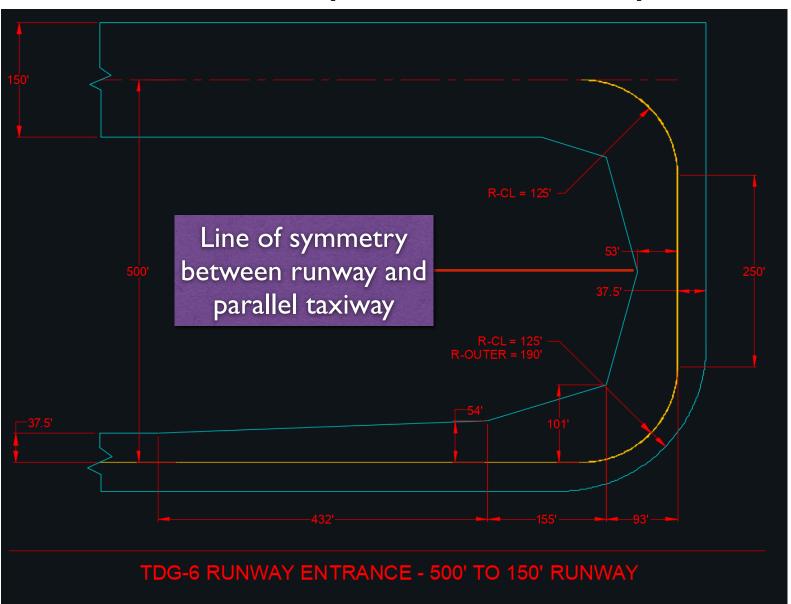
- Use the FAA Taxiway Fillet Design Tool or tables in Appendix J of the Faa AC 150/5300-13B
- If the steering angle estimated by the fillet design tool exceeds 50 degrees, increase R-CI and fillet dimensions according to Table 4-4

Step 3

• Draw the solution using CAD software

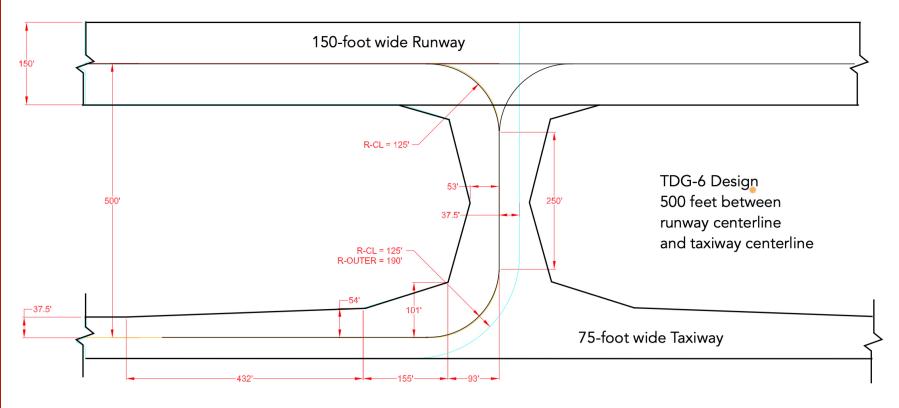
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TDG-6 Runway Entrance Taxiways



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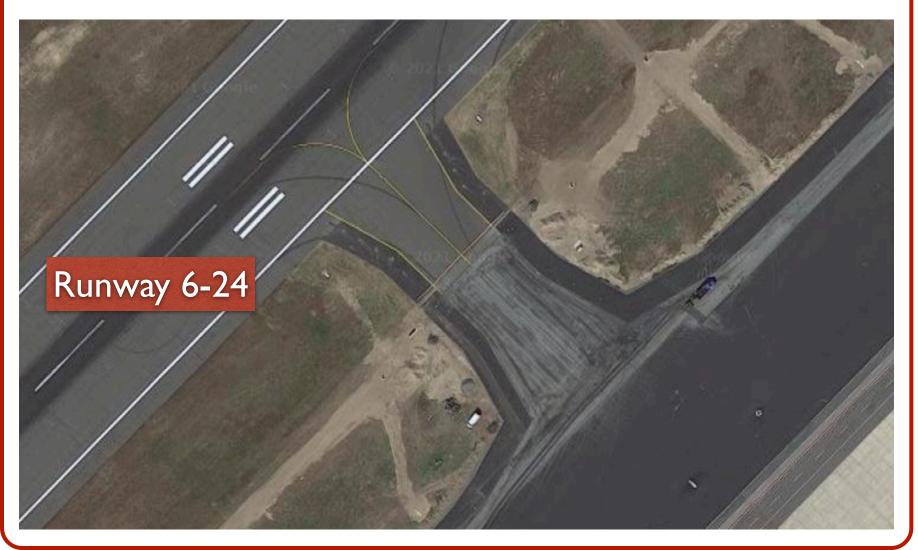
Right Angle Runway Exit (TDG-6 Group) Solution



Build the runway exit as a projection of a runway entrance exit with two sides

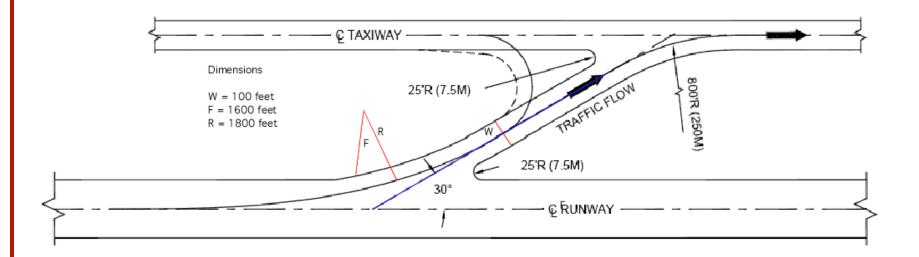


Sample Implementation of a Right Angle Runway Exit (Bradley Intl.Airport)



Acute Angle Runway Exits Also Called High Speed Runway Exits

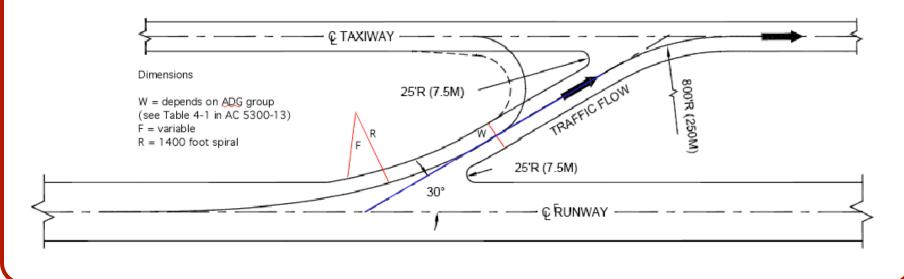
Acute Angle or High-Speed Runway Exit 30 Degree - Constant Radius (**Old Standard** with 1800 foot centerline radius)



Research done in 1960s by Horonjeff demonstrated that a 1,800 foot radius would allow aircraft to exit at 60 knots.

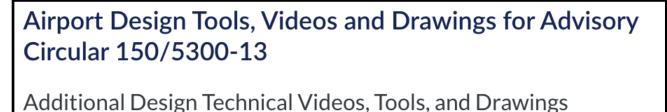
Acute Angle or High-Speed Runway Exit 30 Degree - Spiral Design (Another Legacy High-Speed Runway Exit Design)

- Nominal 1400 feet centerline spiral
- FAA computer program AD42.exe application for design (companion computer program to AC 5300-13)



Current Design Standards

 Available in ready made DXF files at: <u>https://</u> <u>www.faa.gov/airports/engineering/airport_design</u>

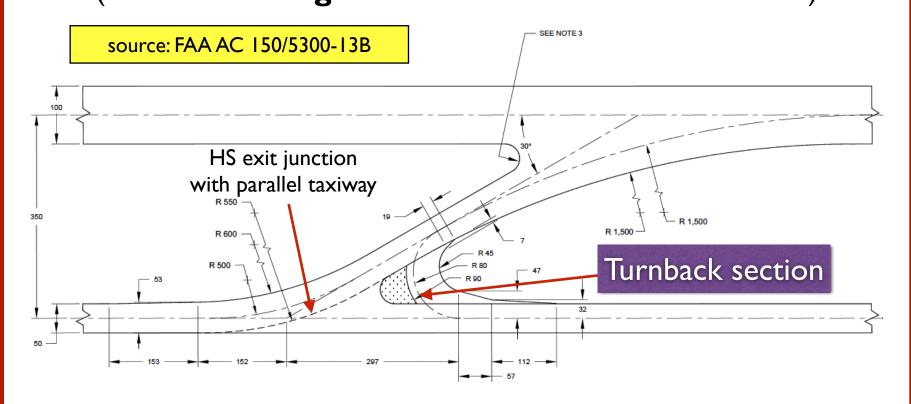


Advisory Circular

- <u>AC 150/5300-13 Airport Design</u>
- <u>Runway Design Matrix Tool</u>

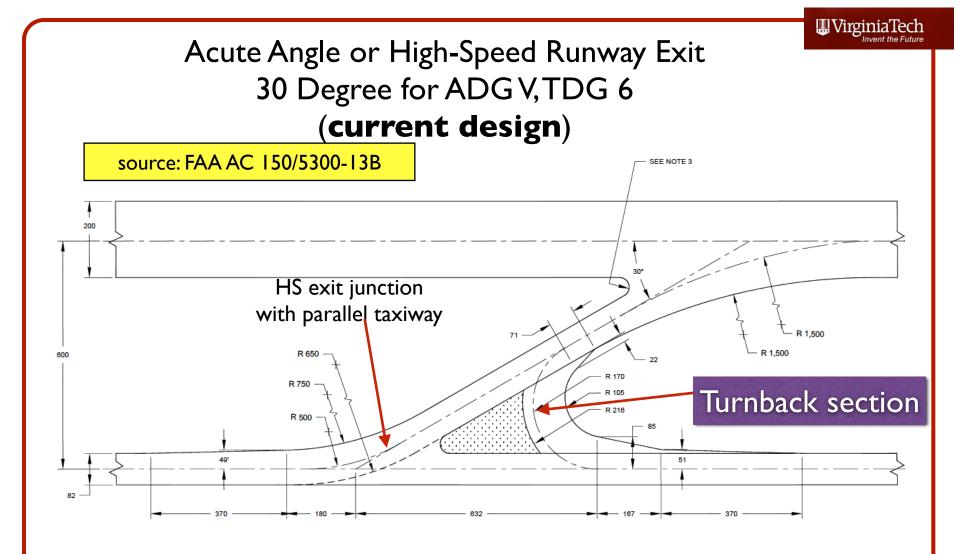
EXPAND ALL	COLLAPSE ALL
Airport Design Too	ols
Drawings by Taxiv	vay Design Group (1

Acute Angle or High-Speed Runway Exit 30 Degree for ADG V, TDG 3 (Current Design - 1500 ft. Centerline Radius)



Note: 350 foot separation between runway centerline and parallel taxiway centerline

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Note: 600 foot separation between runway centerline and parallel taxiway centerline

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🛄 Virginia' Acute Angle or High-Speed Runway Exit for TDG 6 (FAA DXF file) HS exit junction with parallel taxiway -R-1500 37.5'-R-1500 R 800 CONCENTRIC WITH CL Turnback section

HIGH-SPEED EXIT FOR TDG-6 - 500' TO 150' RUNWAY

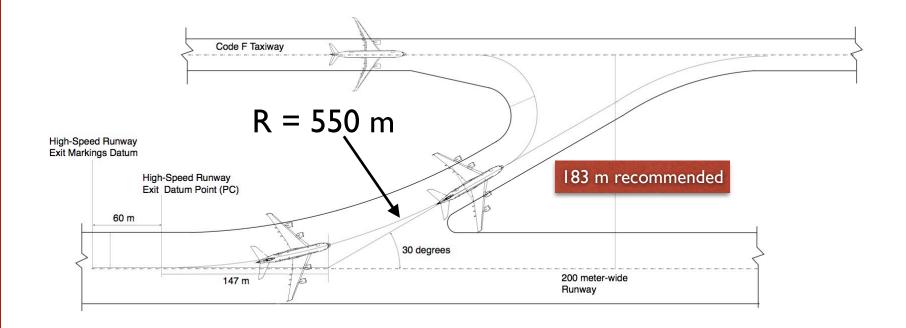
Drawings available at: <u>https://www.faa.gov/airports/</u> <u>engineering/airport_design</u>

Comparison Between HS Exit Designs

- The old 30-degree acute angle exit standard was originally proposed by Horonjeff et al. with a constant centerline radius of 1800 feet
- In the early 1990s, a 1400 foot spiral transition was introduced as the 30 degree (acute angle) design
- In 2013, the FAA went back to a constant radius design (1500 feet at centerline)
- Note that in the current designs suggested by FAA, the transition centerline radii dimensions change at the junction with the parallel taxiway for various TDG groups

ICAO Rapid Exit Taxiways

- ICAO uses a constant radius 550 meter curve
- The ICAO standard is similar to the old FAA standard (first developed by Horonjeff in 1961)



Design and Operational Considerations

- Virginia Tech observations suggest that most HS exits are used 10-20 knots below their design speed (60 knots)
- Perhaps this could be one reason for the FAA to change course
- However, Virginia Tech research suggest that pilots do not like abrupt transitions from a 150 foot runway width to a narrow 75 foot HS runway exit (as is the case for the current FAA design)
- Always be generous with the transition form a wide runway to a narrow runway exit
- HS runway exits are more effective when the separation between the runway and the parallel taxiway is at least 600 feet

To be Effective High-Speed Runway Exits require Good Separation Between Runway and Parallel Taxiway

Table 4-5. Runway to Taxiway Separation for Reverse Turns from a High-Speed Exit Based on TDG

Runway Centerline to	TDG									
Taxiway/ Taxilane Centerline	3	4	5	6						
Recommended separation	350 ft	450 ft	450 ft	600 ft						
	(107 m)	(137 m)	(137 m)	(183 m)						
Radius for 150-degree	79 ft	121 ft	121 ft	152 ft						
turn after 30-degree exit	(24.1 m)	(37 m)	(37 m)	(46 m)						
Minimum separation ¹	348 ft	427 ft	427 ft	485 ft						
	(106 m)	(130 m)	(130 m)	(148 m)						

Note 1: Minimum separation distance based on the standard 30-degree high speed exit and maximum 50degree steering angle for the reverse turn.

Source: Table 4-5 FAA AC 150/5300-13B

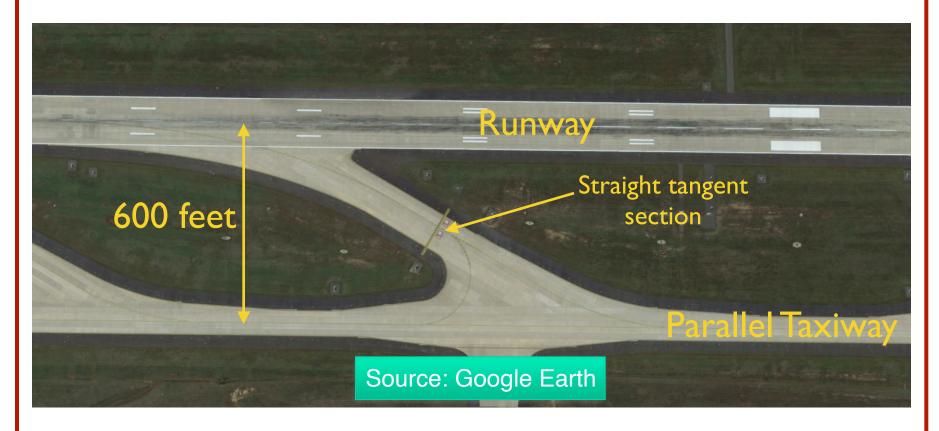
Example of a HS Runway Exit with 400 feet Separation (not recommended)

- Little tangent section on the HS exit for deceleration
- Taxiway is too close to the runway (pilots will exit at lower speeds)

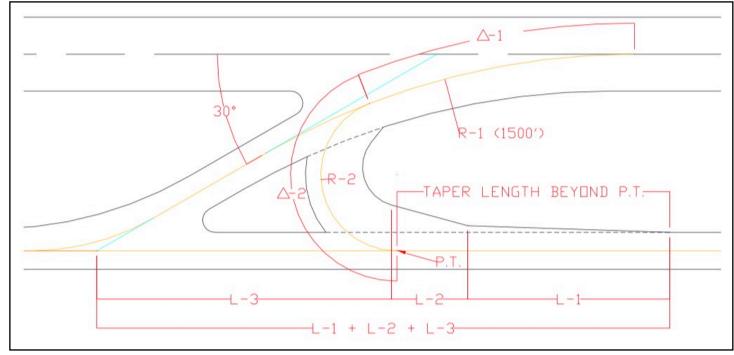


Example of a HS Runway Exit with 600 feet Separation (good practice)

- A generous tangent section on the HS exit for deceleration
- Pilots will exit at higher speeds in such design



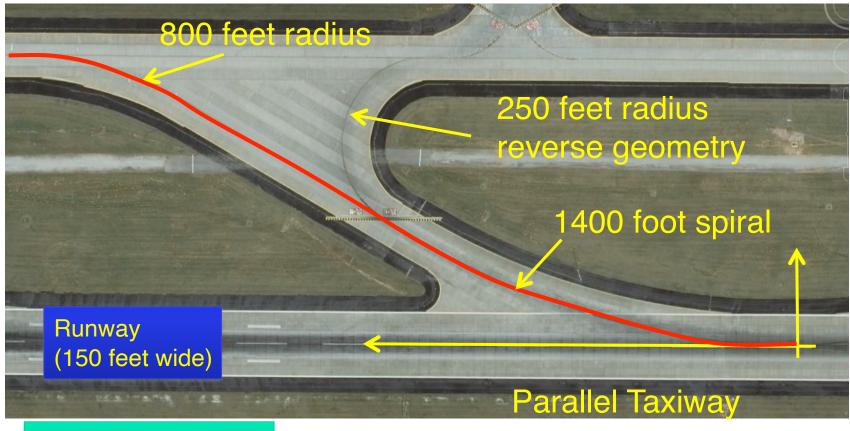
FAA Spreadsheet Calculator for Runway Exit Design



 https://www.faa.gov/airports/engineering/ airport_design/ **Wirginia**

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Example Implementation (ATL) 30 Degree Angle Runway Exit



Source: Google Earth



High-Speed Speed Exits (IAD) (Standard 30 degree angle)



No longer recommended Large pavement overlap can confuse pilots

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Issues with 30 Degree Runway Exits

- The FAA recommends a minimum runway-taxiway separation of 600 feet for High-Speed runway exits
- Some airport have used 30 degree runway exits with only 400 feet between runway and taxiway centerlines (avoid this is bad practice)
 - The result is low exits speeds and possible issues with busting hold lines
- Be careful and try to provide the minimum 600 foot recommended distance
- Consider limited pilot visibility while crossing active runways

FAA Airport Design Website

 Available at: <u>https://www.faa.gov/airports/</u> <u>engineering/airport_design/</u>

Available tools:

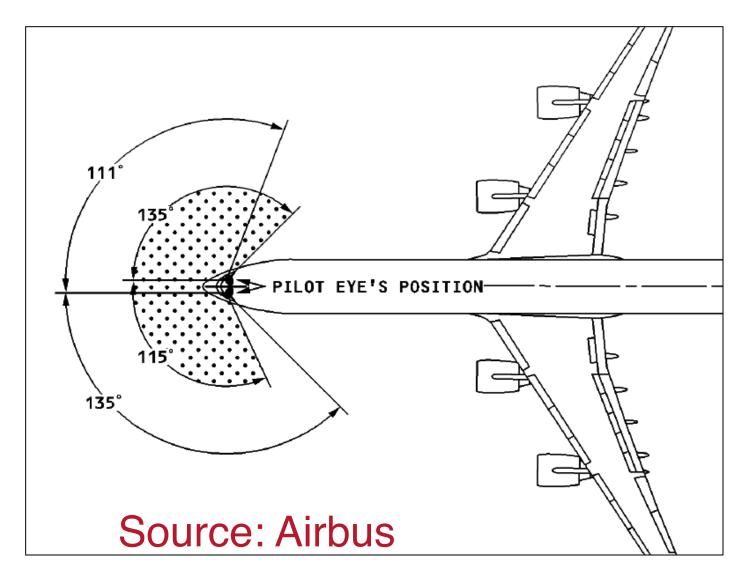
FAA Microsoft Excel Spreadsheet Programs

- Taxiway Design Tool for High Speed Exits (MS Excel)
 - Instructions for how to design High Speed Exits (PDF)
- Taxiway Fillet Design Tool (MS Excel) (added 6/6/2018)
 - Taxiway Fillet Design Tool User's Guide (PDF)

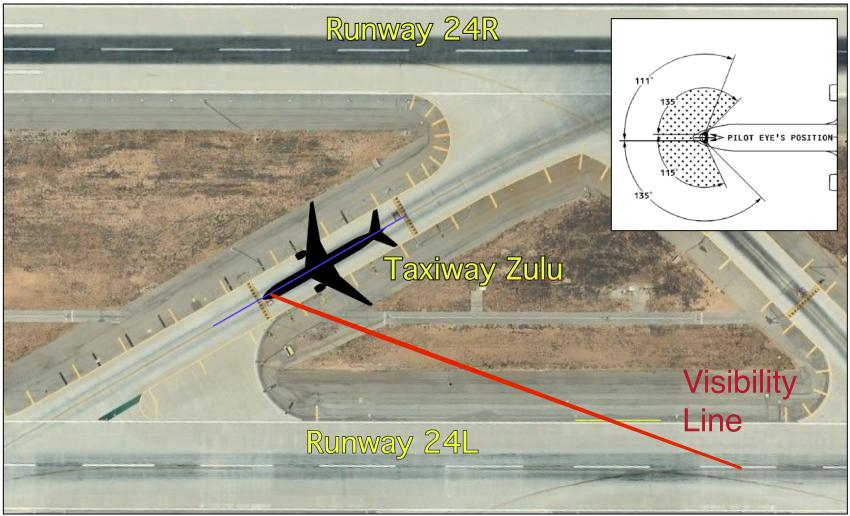
Tool developed at Virginia Tech

 Runway Exit Design Interactive Model (REDIM) ^[] (added 5/6/2020) REDIM is a computer model developed to locate and design highspeed runway and right angle exits at airports. The model uses kinematic equations to characterize the aircraft landing dynamics

Airbus A340-600 Visibility from Cockpit



Sample Limited Visibility due to High-Speed Runway Exits (LAX Airport)



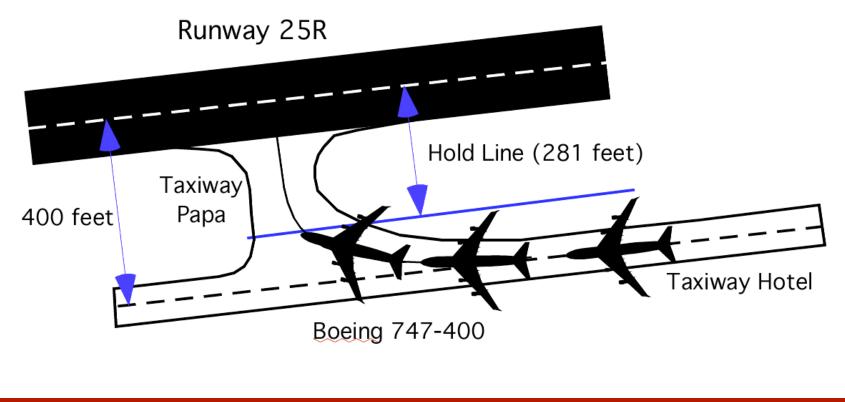
Final turning angle at hold line = 30 degrees

UirginiaTech

UrginiaTech Example of Limited Visibility due to Short Runway-Taxiway Distance **Runway Centerline** Airbus A340-600 visibility limit Holding Line (281 feet) нтАч **GUIDELINE** YAWIXAT 400 feet (18.4) R 25.9 (R 85.0) (53.6) 9.0 2.2 Center Taxiway NLG PATH COIDERINE PARALLEL TO COCKPIT TRACK T32330 JANIMON Airbus 340-600 (2.6.2) 0.8 turns 39 degrees at hold line (0.27) 55.9

Example of Limited Visibility from Aircraft Cockpit Driven by Hold Line Location

• Before the aircraft nose reaches the hold line, the aircraft wingtip violates the hold line distance



Runway Exit Placement

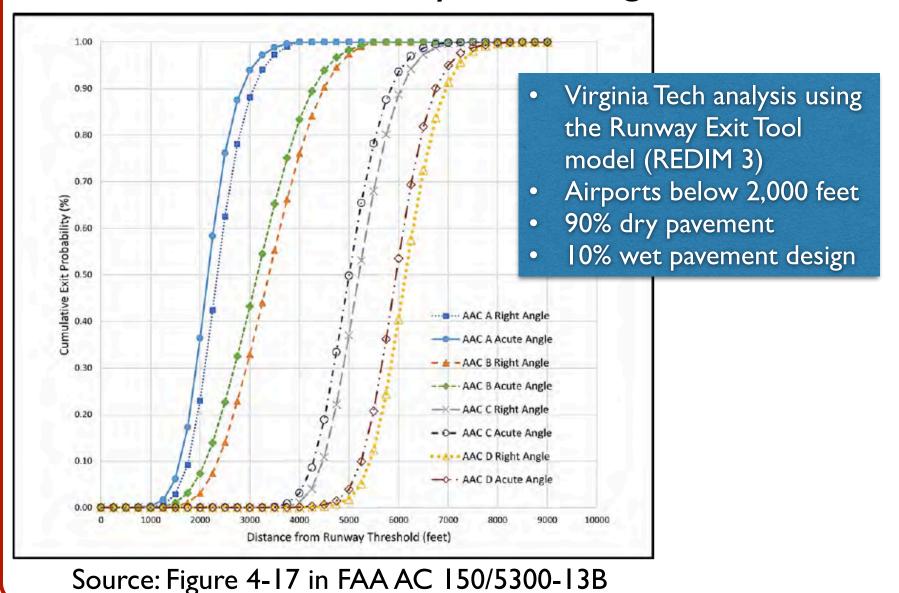
Procedures to Locate Runway Exits

- Factors that affect the runway exit locations:
 - Fleet mix
 - Operations/hr
 - Environmental conditions (wet vs. dry pavement)
 - Terminal or gate locations
 - Type and number of runway exits
- Manual tables developed by ICAO and FAA
- Use computer models like REDIM Runway Exit Design Interactive Model (Developed at Virginia Tech for the FAA and NASA)

Methods to Locate Runway Exits

- Graphical solution of cumulative exit probability runway exit use charts (see Figure 4-17 in AC 150 5300-13B)
- Three segment method using simple aircraft kinematics
- Use the Runway Exit Design Tool (REDIM version
 3) developed by Virginia Tech for the FAA

Cumulative Runway Exit Design Curves

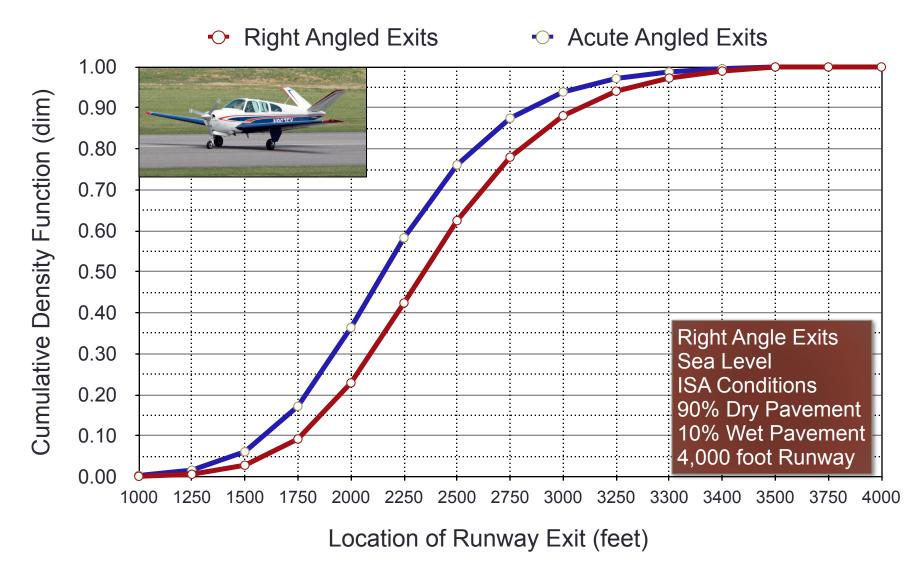


Airport Planning and Design (Antonio A. Trani)

Invent the Future

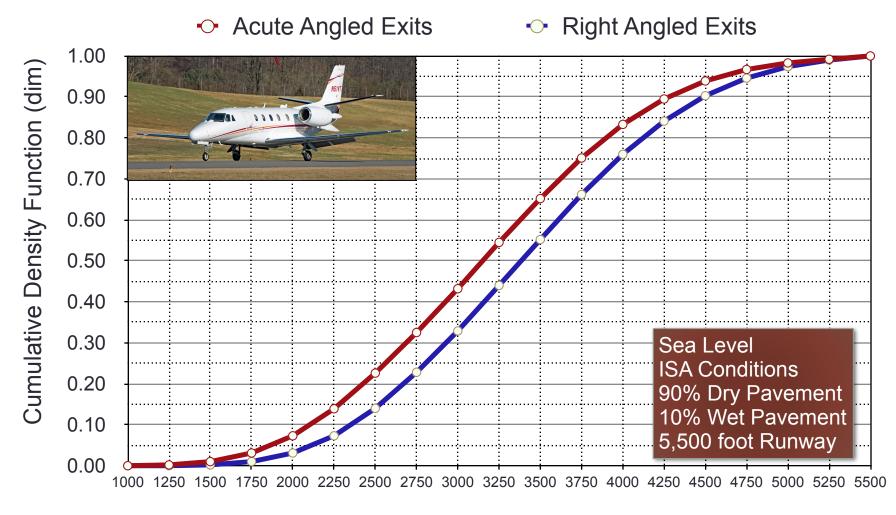
VirginiaTech

Runway Exit Design for AAC A Class Aircraft



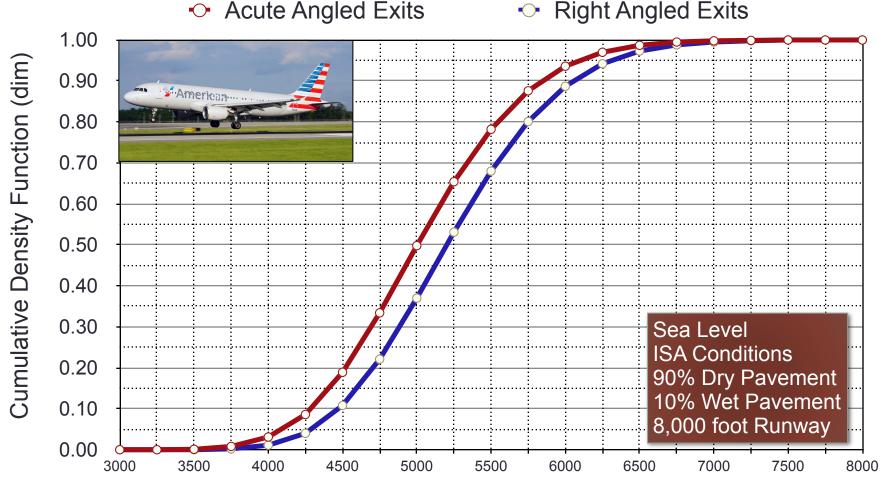
Air Transportation Systems Laboratory

Runway Exit Design for AAC B Class Aircraft



Air Transportation Systems Laboratory

Runway Exit Design for AAC C Class Aircraft

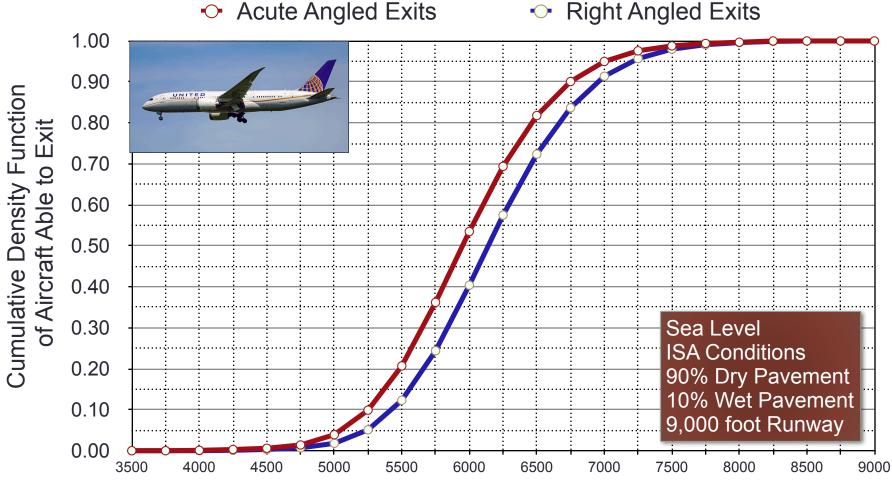


VirginiaTech Invent the Future For Typical Acute Angle Exit Locations, Increase Runway Exit Location by 150 feet for Every 1,000 ft Airfield Elevation Change Sea Level • 3,000 ft 6,000 ft 1.00 0.90 **Cumulative Density Function** 0.80 of Aircraft Able to Exit 0.70 0.60 0.50 0.40 0.30 Acute Angled Exits 0.20 90% Dry Pavement 10% Wet Pavement 0.10 9,000 foot Runway 0.00 3500 4000 4500 5000 5500 6000 6500 7000 7500 8000 8500 9000

Air Transportation Systems Laboratory

VirginiaTech

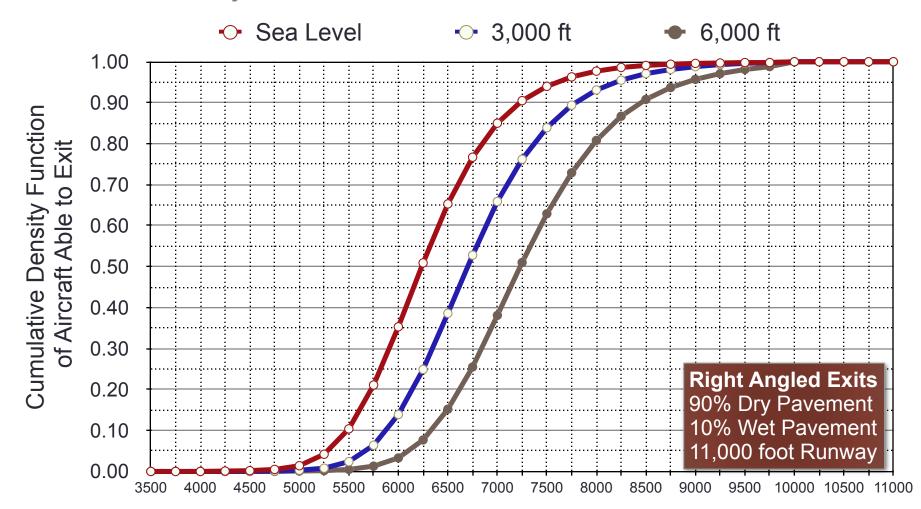
Runway Exit Design for AAC D Class Aircraft



VirginiaTech

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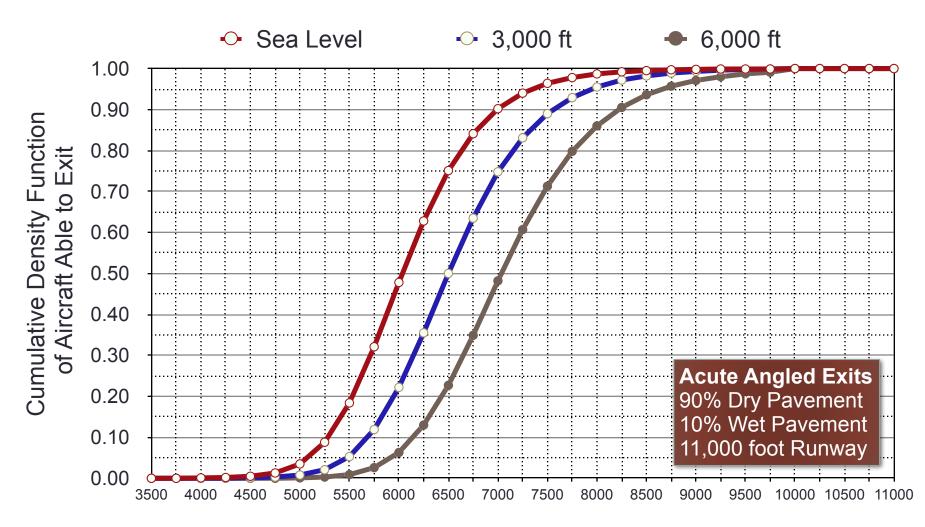
For Typical Acute Angle Exit Locations, Increase Runway Exit Location by 167 feet for Each 1,000 ft Airfield Elevation



Air Transportation Systems Laboratory

Invent the Future

For Typical Acute Angle Exit Locations, Increase Runway Exit Location by 185 feet for Each 1,000 ft Airfield Elevation



Air Transportation Systems Laboratory

Runway Exit Location Example Problem Using Cumulative Runway Exit Design Curves

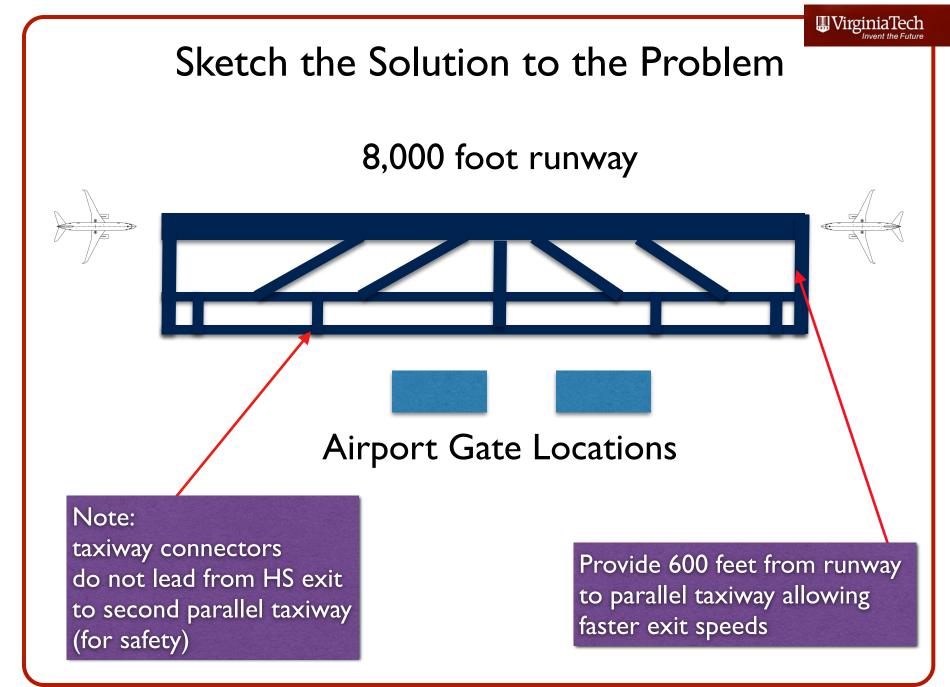


Aircraft and Airport Environment

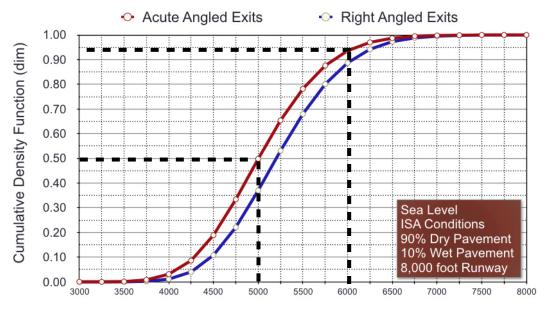
- Airport located at sea level conditions
- 8000-foot runway
- Minimum of 4 runway exits are desired
- AAC groups
 - B 20% of fleet mix
 - C 80% of fleet mix
- More than 30 operations per hour in the peak period

Typical Questions and Workflow

- Locate two high-speed (acute angle) runway exits per direction to accommodate 50% and 95% of AAC C class aircraft
- Sketch a runway exit solution
- Measure ROT using the FAA/Virginia Tech Runway Exit Design Model (REDIM)



Evaluate Runway Exit Usage at Desired Locations Solution to the Problem



Location of Runway Exit (feet)

AAC C Class Aircraft

Location of Two High-Speed Exits Ist - 5,000 feet (50%) 2nd - 6,000 feet (95%) Always use 800 feet as the minimum distance between high-speed runway exits

The criteria is satisfied (1,000 feet between two HS exits)





FAA/Virginia Tech Runway Exit Design Tool REDIM Model



Runway Exit Design Tool (REDIM 4 Web Site)





REDIM

Version 4.0.2 - Date: 12/20/2022

Virginia Tech - Air Transportation Systems Lab

Dr. Antonio Trani (Team Leader) Nicolas Hinze (Team Co-Leader) Navid Mirmohammadsadeghi Mani Bhargava Reddy Bollempalli Mihir Rimjha Arman Izadi Afshin Olamai Armin Zolfaghari

FAA - Project Sponsors

Kent DuffyFAA Airports Planning and Environmental Division (APP-400)Lauren VitaglianoFAA William J. Hughes Technical CenterChristina NuttingFAA Airports Planning and Environmental Division (APP-400)

Web site to download the REDIM model

https://atsl.cee.vt.edu/products/runwayexit-design-interactive-model-redim-1.html Runway Exit Design Interactive Model V4 (REDIM-V4)

Explore

Runway Exit Design Interactive Model V4 (REDIM-V4)

84 26L-8

Download REDIM 4

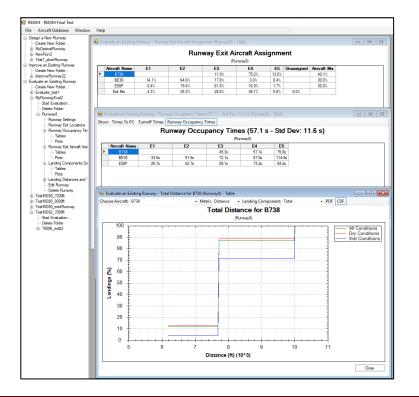
- REDIM 4.0.2 Windows Installer
- User Group
- User Manual
- FAQs
- Change Log





General Information About the Model

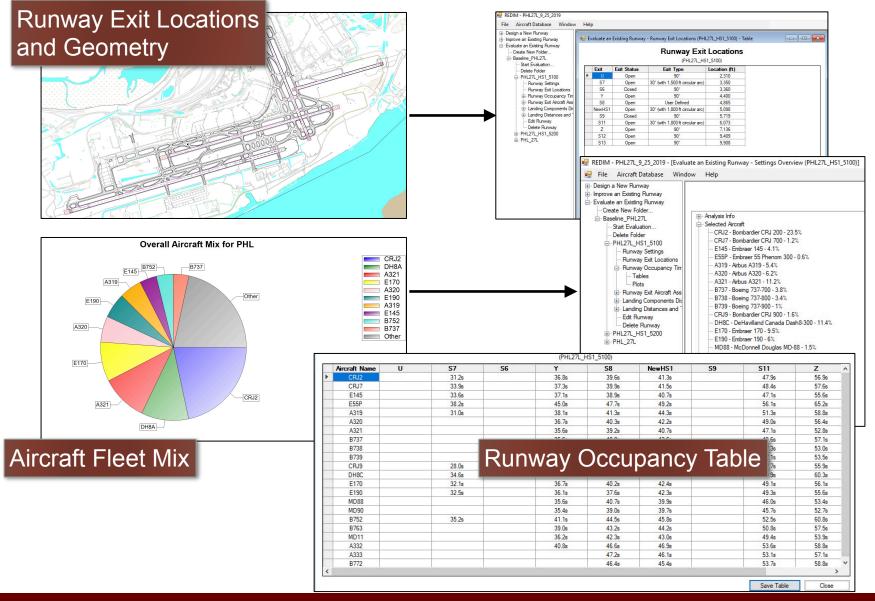
- Model has three analysis modules:
 - a) Evaluation of an existing runway
 - b) Improvements to an existing runway
 - c) Design optimal locations for a new runway



Model uses Monte Carlo Simulation to predict aircraft landing roll performance

- Stand-alone Windows
 application
- Requires ~1.8 Gb of hard disk space
- New runway clustering
- Improvements to landing roll profile calculations

WirginiaTech Invent the Future Runway Exit Design Model Workflow



Air Transportation Systems Laboratory





Runway Exit Design Tool Outputs

Analysis	Purpose	Outputs Produced		
Aircraft Mix	Provides an overview of aircraft fleet mix	Percent of aircraft types simulated in the analysis		
Runway Occupancy Time	Provides three values of runway occupancy time measured at two locations:1.Fuselage out2.At hold bar	 1.Average ROT (in seconds) by runway exit and aircraft (table format 2.Average ROT (in seconds) by runway exit and aircraft (graphical format) 3. Weighted average ROT for the complete aircraft mix using the runway 4. Standard deviation of ROT for the complete fleet mix 5. Individual landing roll times for every aircraft simulated by the model (~50,000 landings per aircraft) 		
Runway Exit Utilization	Provides information about aircraft assigned to each exit	 Percent of individual aircraft assigned to each runway exit Individual ROT by aircraft and runway exit 		
Aircraft Landing Performance	Provides individual landing event information (REDIM uses a Monte Carlo Simulation Process)	 Landing roll distributions (CDF and PDF) by runway condition (wet or dry) in table format Landing roll distributions (CDF and PDF) by runway condition (wet or dry) in graphical form Landing roll distances and times by aircraft and runway pavement condition (wet or dry) a) Air distance and air time (time to nose gear touchdown) b) Nominal braking distance and time c) Extra roll distance and time d) Turnoff distance and time 		



Differences with Previous Runway Exit Model

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Item	Older Model (REDIM 2.1)	New Model (REDIM 3)		
Wind information	Single wind speed and direction	Landing observations are effected by local wind conditions		
	Airports have complex wind patterns	Landing events database has wind speed and direction for each landing		
		REDIM 3 designs for average wind conditions included in the landing speed distributions collected at 37 ASDE-X airports		
Runway gradient	Ten values of local gradients along the runway Model will calculate the average gradient and apply a very small correction factor	Runways designed for commercial operations have limited gradients by regulation (0.8% in first quarter of runway and a maximum of 1.5%)		
	apply a very small concetion factor	The correction factor for such runways is very small		
D (1'4'		We plan to investigate this issue in the future		
Pavement conditions	50/50 wet dry default condition	10/90 default condition in new model		
	Wet pavement conditions reduce nominal deceleration	Rainfall data collected at selected airports provided the basis for the new default		
		Wet pavement conditions reduce nominal deceleration		
Safety factor	Turnoff safety factor (user defined)	The new model is based on extensive runway and runway exit data		
		Observed runway exit speeds in the new model make the use of a safety factor in the turning maneuver unnecessary		
Aircraft fleet	5 aircraft modeled directly (Douglas DC9-30,	298 aircraft modeled		
	McDonnell Douglas MD-80, Boeing 727-200, Boeing 737-300 and Boeing 757-200)	Aircraft performance adjusted for airport elevation and temperature		
	70+ aircraft modeled indirectly based on landing distance parameters adjusted for airport elevation	Landing roll distributions use Kernel Density Functions (KDE) for individual aircraft (functions of runway length and runway exit types)		
	Aircraft performance adjusted for airport elevation and temperature	Model defaults to Aircraft Approach Group (AAC) category if a Kernel distribution does not exists for the aircraft in question		
	Model assumed all landing roll distributions to be normally distributed truncated to 2.5 sigma			





Runway Exit Model 4 : Aircraft Database

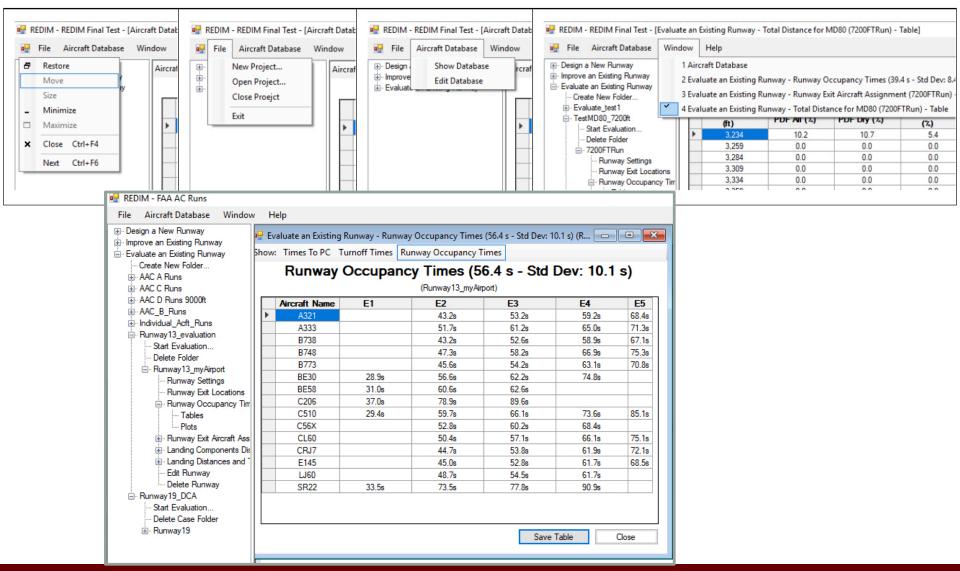
- 330 aircraft modeled (directly or indirectly)
- Improved database consistent with the updated FAA Aircraft Characteristics Database (ACD)
- Includes the latest generation of aircraft (Airbus 220-300, A320neo, Boeing 737-8Max, etc.)

	ADG III Aircraft							
Aircraft ID	Aircraft Name	Engine Type	Aircraft Design Grou	Aircraft Design G	roup (ADG): V	Ness	Coor	
A19N	Airbus A319 Neo	Jet	III	5				
A20N	Airbus A320 Neo	Jet	III					
A21N	Airbus A321 Neo	Jet	III			Casina	Aircraft	
A318	Airbus A318	Jet	Ш	Aircraft ID	Aircraft Name	Engine Type	Design Group	
A319	Airbus A319	Jet	III	A332	Airbus A330-200	Jet	V	
A320	Airbus A320	Jet	III	A333	Airbus A330-300	Jet	V	
A321	Airbus A321	Jet	III	A337	Airbus A330-700 - Beluga XL	Jet	V	
AT42	Aeropatiale ATR-42-200	Turboprop	III	A338	Airbus A330-700 - Beluga AL	Jet	V	
AT43	Aeropatiale ATR-42-300	Turboprop	III				-	
AT44	Aeropatiale ATR-42-400	Turboprop	III	A339	Airbus A330-900	Jet	V	
AT45	Aeropatiale ATR-42-500	Turboprop	III	A342	Airbus A340-200	Jet	V	
AT46	Aeropatiale ATR-42-600	Turboprop	111	A343	Airbus A340-300	Jet	V	
AT71	Aeropatiale ATR-72-100	Turboprop	III	A346	Airbus A340-600	Jet	V	
AT72	Aeropatiale ATR-72-200	Turboprop	III	A359	Airbus A350-900	Jet	V	
AT73	Aeropatiale ATR-72-300	Turboprop	III	B742	Boeing 747-200	Jet	V	
AT74	Aeropatiale ATR-72-400	Turboprop	111	B744	Boeing 747-400	Jet	V	
AT75	Aeropatiale ATR-72-500	Turboprop	III	B772	Boeing 777-200	Jet	V	
AT76	Aeropatiale ATR-72-600	Turboprop	III	B773	Boeing 777-300	Jet	V	
B37M	Boeing 737 MAX 7	Jet	III	B77L	Boeing 777-200LR	Jet	V	
B38M	Boeing 737 MAX 8	Jet	III	B77U	Boeing 777-300ER	Jet	V	
B39M	Boeing 737 MAX 9	Jet	III	B77W	Boeing 777-300ER Boeing 787-8	Jet	V	

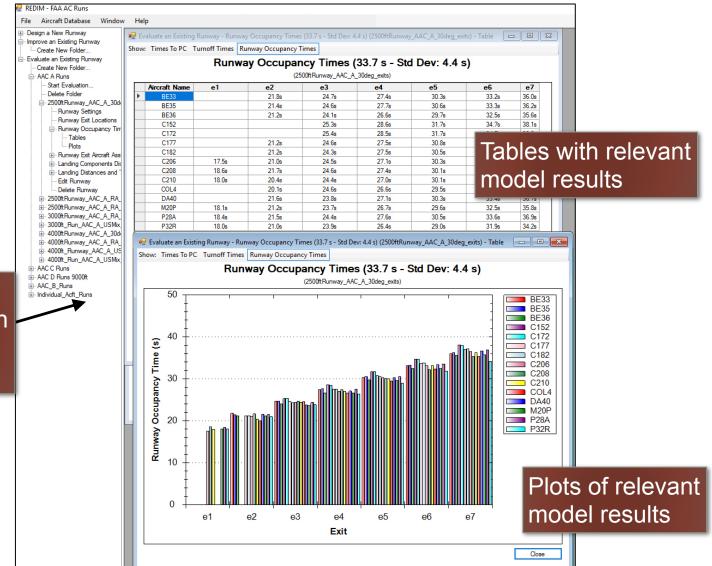
VirginiaTech



REDIM 4 Menu Structure



VirginiaTech Invent the Future Interface and Panels in the Runway Exit Design Model



Navigation and project panel with information and results

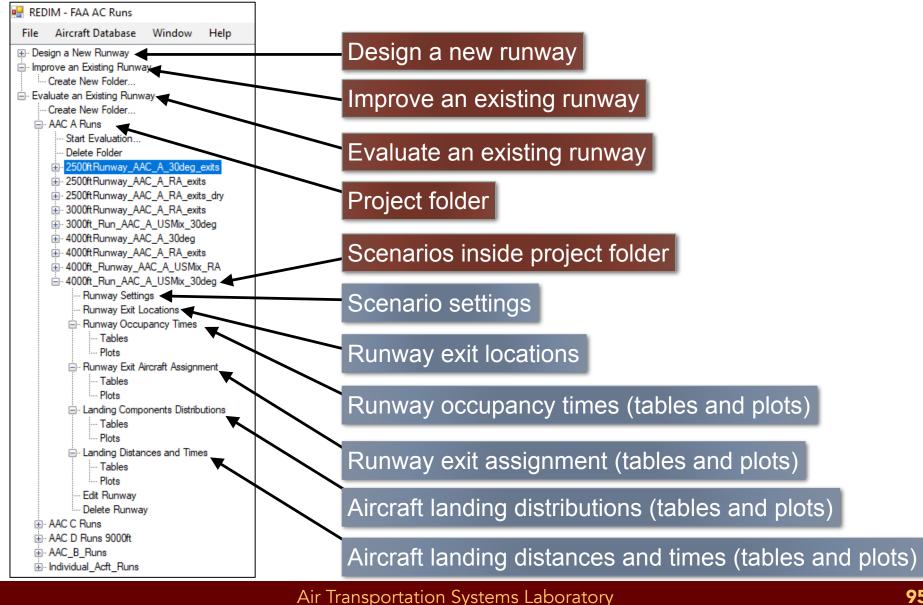
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Navigation/Project Panel Hierarchy

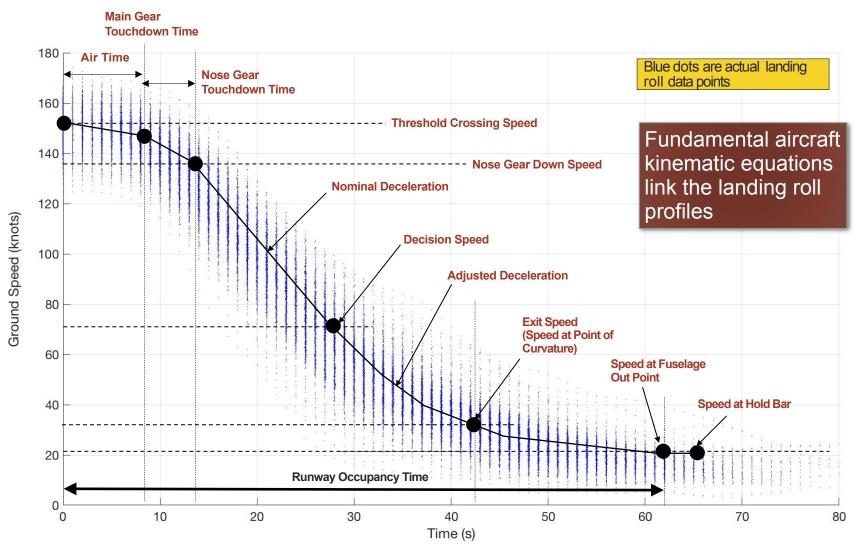




Runway Exit Model Landing Roll Profile Phases Modeled

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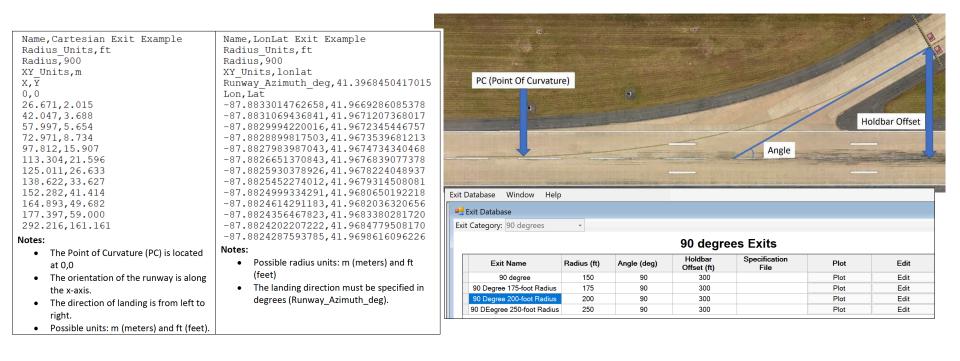


Runway Exit Model 4 Improvements: Runway Exit Data Handling

REDIM offers default runway exits

irginiaTech

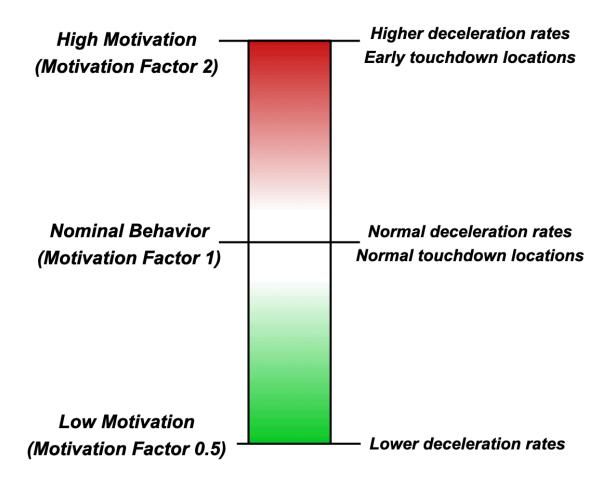
- REDIM offers a detailed procedure to define custom runway exits using relative(x-y) or absolute coordinates (latitudelongitude)
- REDIM 4 can store runway exits in a runway exit database file





Pilot Motivation Factor

- We developed methods to characterize pilot motivation through statistical analyses of individual aircraft data
 - Deceleration rates
 - Touchdown distances
- Briefed a group of pilots invited by the FAA to understand factors that lead to pilot motivation
- All motivation factors provided in the model are within the kinematic capabilities of each aircraft







Motivation Factor in the REDIM 4 Model Interface

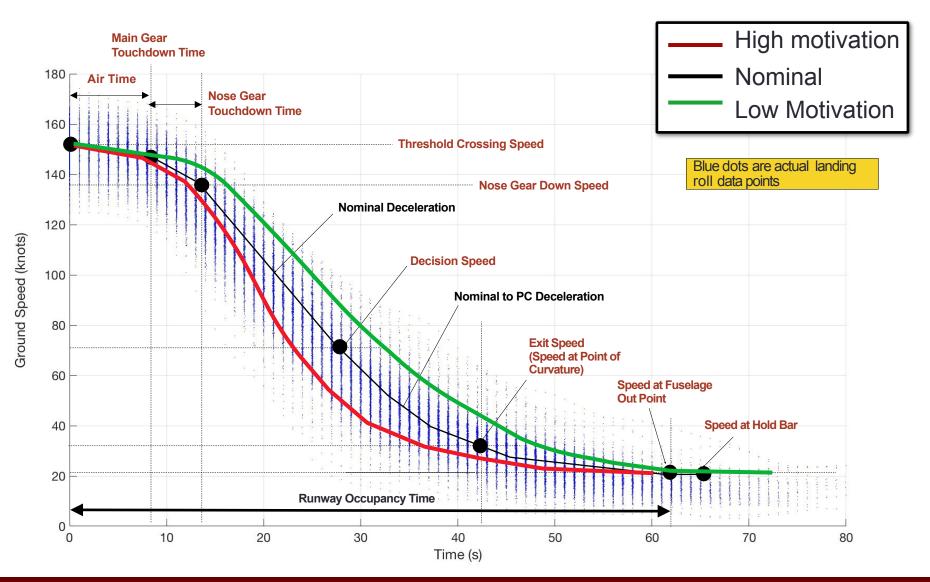
Evaluate an Existing Runway - Step 2 - Define Aircraft Mix for New Runway

Aircraft ID	Aircraft Name	Aircraft Design ▲ Group	Aircraft Approach Category	Aircraft Mix (%)	Motivation Factor	^
AT76	Aeropatiale ATR-72-600	III	В		1	
B37M	Boeing 737 MAX 7	III	С		1	
B38M	Boeing 737 MAX 8	III	D	25	2	
B39M	Boeing 737 MAX 9	III	D		1	
B712	Boeing 717-200	III	С		1	
B721	Boeing 727-100	III	С		1	
B722	Boeing 727-200	III	С		1	
B733	Boeing 737-300	III	С	25	2	
B734	Boeing 737-400	III	С		1	
B735	Boeing 737-500	III	С		1	
B736	Boeing 737-600	III	С		1	
B737	Boeing 737-700	III	С	25	2	
B738	Boeing 737-800	III	D	25	2	
B739	Boeing 737-900	III	D		1	
BCS1	Airbus A220-100	III	С		1	
C55B	Cessna Citation Bravo	III	В		1	
CRJ9	Bombardier CRJ 900	III	С		1	-
DC91	Douglas DC-9-10	III	С		1	
DC93	Douglas DC-9-30	III	С		1	1
DH8B	DeHavilland Canada Dash8-200	III	В		1	
DH8C	DeHavilland Canada Dash8-300	III	В		1	

During the landing simulation, Boeing 737-700 will use a motivation factor of 2.0 (high motivation) VirginiaTech



Landing Roll Profiles versus Pilot Motivation



Air Transportation Systems Laboratory

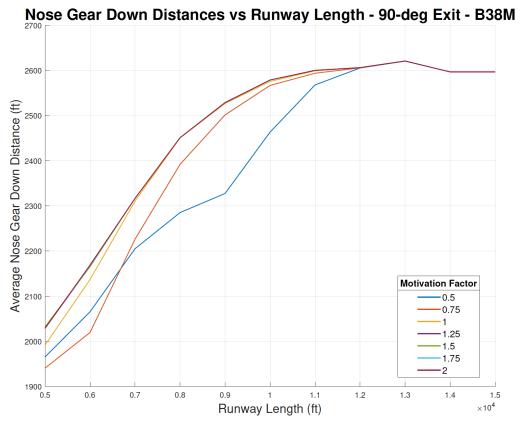
Pilot Motivation in the REDIM 4 Model Starts with Modified Touchdown Locations

 The natural trend of touchdown locations is built into the model based on actual data

inia'l'ech

- Short runways produce early touchdown locations
- High motivation factors above 1.5 may reduce the touchdown location by 120-150 feet (~4.8%) on a 8,000-foot runway

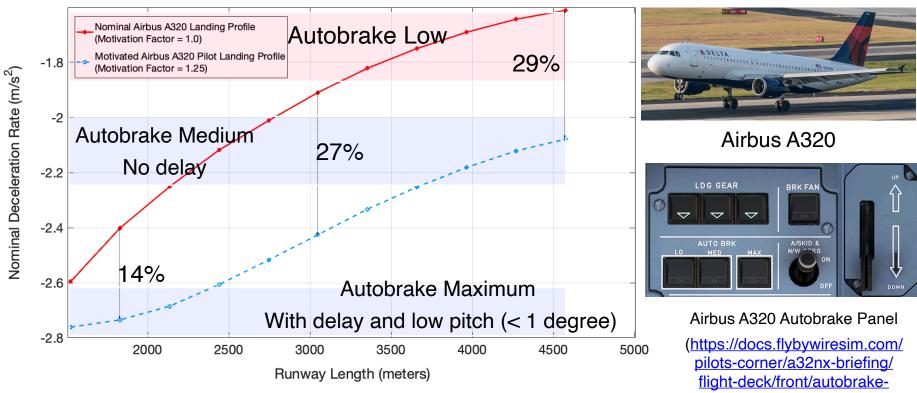








Pilot Motivation Factor Effect on Nominal Deceleration Rate Spread Between MF 1 and MF 1.25 Averages 20% for Runways Up to 3,050 meters (10,000 feet)

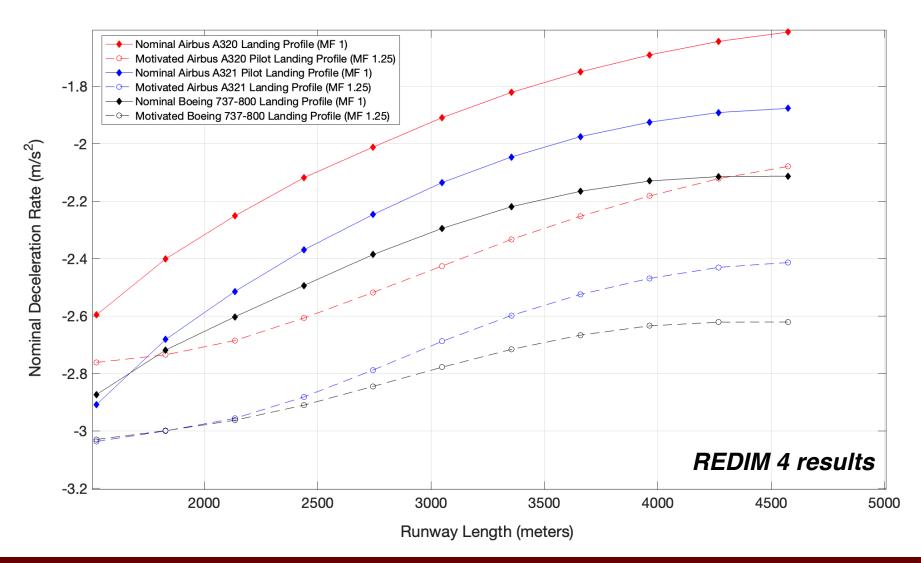


<u>gear/</u>)

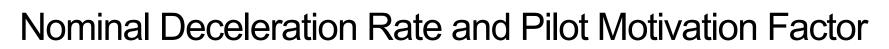


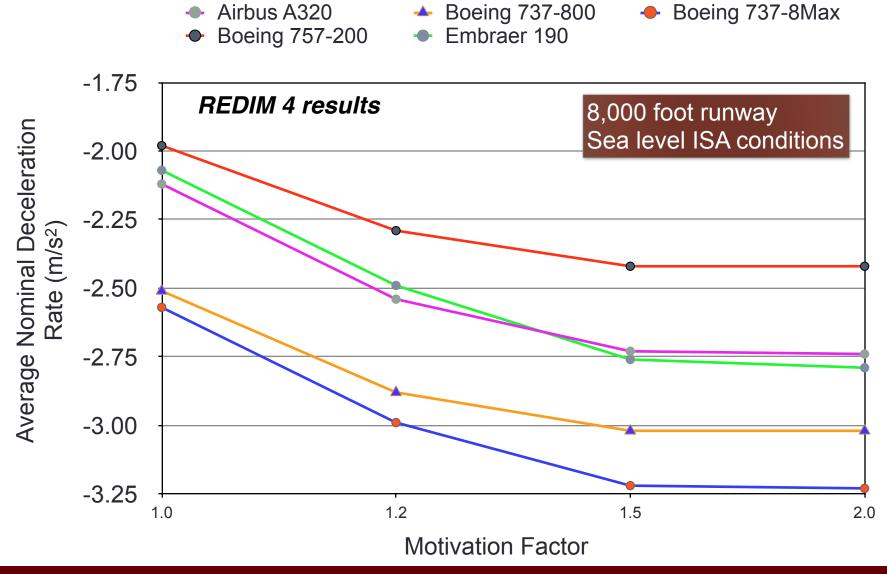
Nominal Deceleration Rate and Pilot Motivation Factor

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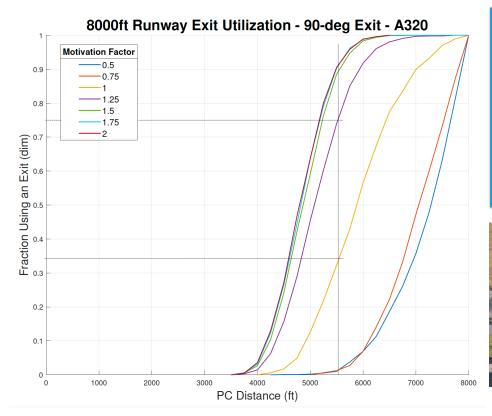


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Practical Implications of Changing the Pilot Motivation Factor on a 2,440-meter (8,000 feet) Runway



Increasing the pilot motivation factor from 1.0 to 1.25 doubles the cumulative runway exit probability of right-angle exits at a location 5,500 feet along the runway.



Airbus A320





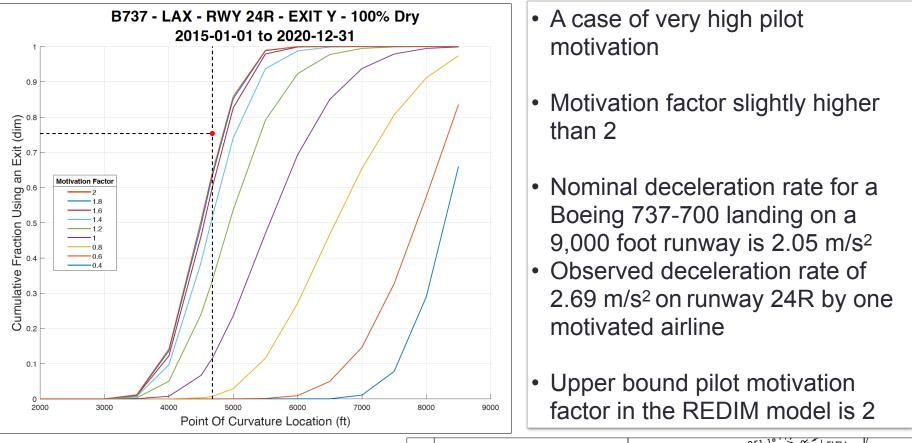
Example Motivation Factors at Selected Airports

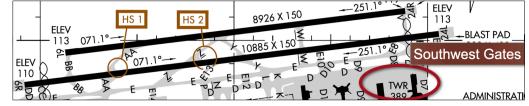
- Practical examples of how to translate motivation factor (MF) to real-world landing performance
 - Los Angeles Runway 24R
 - LaGuardia Runway 31
- REDIM model motivation factor guidance for model users





77.5% of Motivated Pilots of Boeing 737-700 Landings Exit at Yankee (Y)

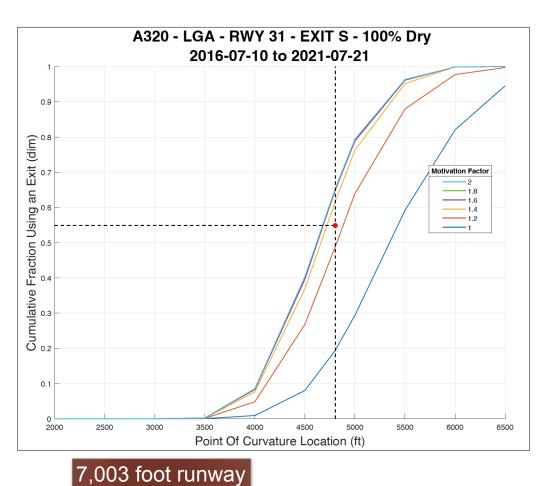






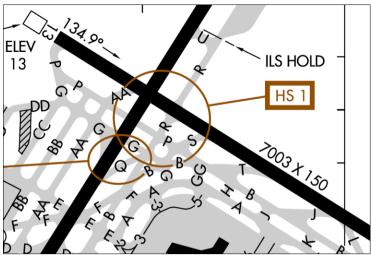


55% of Airbus A320 Landings at LGA Runway 31 Use Exit Sierra (S)



2017-2020 Data

- A case of higher than nominal pilot motivation
- Pilots are motivated is to avoid crossing runway 4/22 while landing on runway 31
- Observed motivation factor is ~1.25

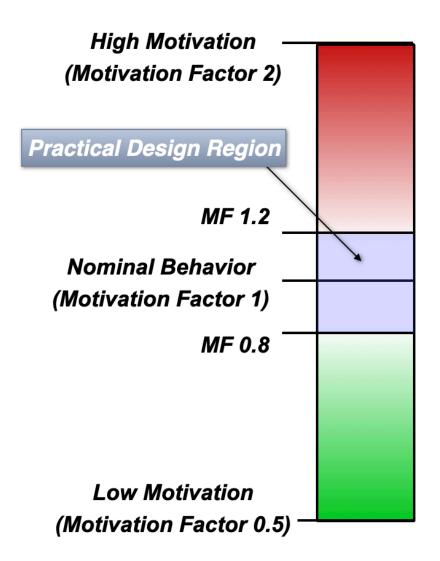






Recommended Guidance for Motivation Factors

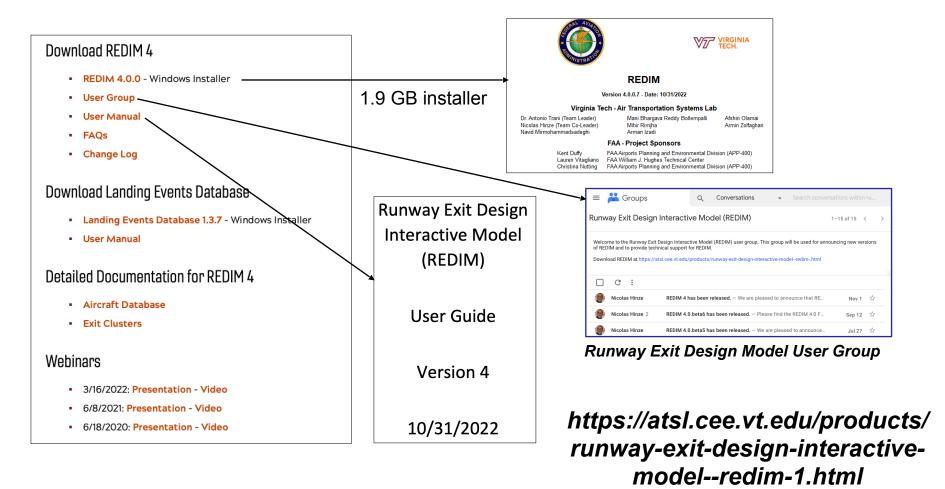
- REDIM 4 is designed to predict nominal landing roll behavior on a runway with a motivation factor MF 1.0
- High motivation factors are observed at some U.S. airports (MF 1.25 or higher)
- Practical design guidance should limit the placement of runway exits using MF factors between 0.8 and 1.2 to avoid high deceleration rates that may not be desirable in real-world commercial operations
 - Higher maintenance costs due to heavy braking
 - Passenger comfort
- For narrow-body aircraft, a motivation factor of 1.2 translates into 15-20% increase in deceleration rates compared to the nominal landing conditions







Runway Exit Design Tool (REDIM Model) Resources







Example Use of the REDIM Model Runway Evaluation

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Runway Evaluation Scenario: Fleet Mix

Aircraft	Fleet Mix (%)	Picture
Cirrus SR22	10	N874DC
Cessna Latitude (C680A)	10	
Airbus A320	25	DESTR.
Boeing 737-800	35	
Boeing 787-8	20	



Runway Evaluation Scenario: Runway Exits

Runway Exit	Location (feet)	Exit Type		
Alpha	1500	Right-angle		
Bravo	3000	Right-angle		
Charlie	5000	Acute-angle (R-CL 1500 feet)		
Delta	6200	Acute-angle (R-CL 1500 feet)		
Echo	8000	Right-angle		
Foxtrot	8900	Right-angle		

Runway length = 9000 feet Airport elevation = 3,200 feet Design temperature = 80 deg. F. 90% dry pavement, 10% wet pavement

VirginiaTech

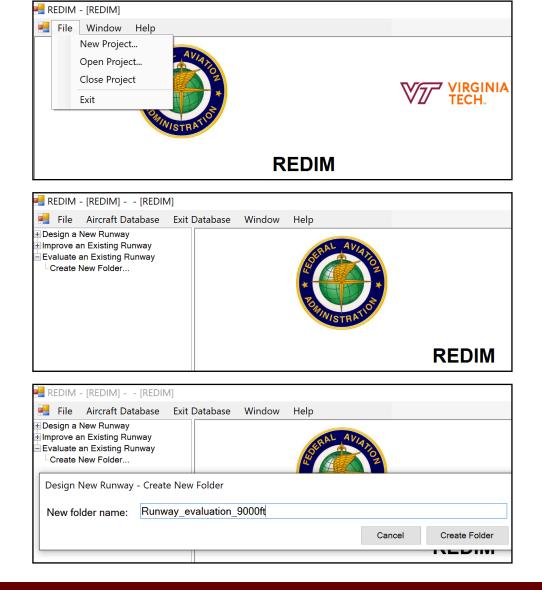
Invent the Future Step by Step Instructions (1)

Step 1: Create a new project

'irginiaTech

Step 2: Evaluate an Existing Runway

Step 3: Create a new folder name





Invent the Future Step by Step Instructions (2)

Step 4: Start the runway evaluation

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Step 5: Define airport conditions



Evaluate an Existing Runway - Step 1 - General Information					
	Step 1: General Information				
Units	Units Runway Occupancy Type				
O Metric	◯ Metric				
Runway In	Runway Information				
Name:	Runway24_ev	aluation			
Length:	2743 meter	S.	9000	feet.	
Width:	46 meter	'S.	150	feet.	
Airport Information					
Elevation:	975 me	ters.	3200	feet.	
Temperature	15 deg	grees Celsius.	80	degrees Fahrenheit.	
Wet Condition	Wet Conditions: 0% 100%.				
10%					



Step by Step Instructions (3)

Evaluate an Existing Runway - Step 2 - Define Aircraft Mix for New Runway

Step 2: Define Aircraft Mix for New Runway

Only provide the aircraft mix for the left or right side of the runway you are modeling.

Aircraft ID	Aircraft Name	Aircraft Design ▲ Group	Aircraft Approach Category	Aircraft Mix (%)
B737	Boeing 737-700	III	С	
B738	Boeing 737-800	III	D	35
B739	Boeing 737-900	III	D	
B73Q	Boeing 737-200 Quick Conversion	III	С	
BCS1	Airbus A220-100	III	С	

Evaluate an Existing Runway - Step 3 - Exits

		Step 3: E	xi	ts
	Only specify exits	s on the left or right side o	of t	he runv
Name Po	oint Of Curvature Location (ft)	Geometry		Open
Alpha 150	500	90 degree	\sim	\checkmark
Bravo 300	000	90 degree	\sim	\checkmark
Charlie 500	000	30 degree High Speed	\sim	\checkmark
Delta 620	200	30 degree High Speed	\sim	\checkmark
Echo 800	000	90 degree	\sim	\checkmark
Echo 800	900	90 degree	\sim	\checkmark
*			\sim	



Step 6: Define the fleet mix Only enter aircraft that are part of the fleet mix

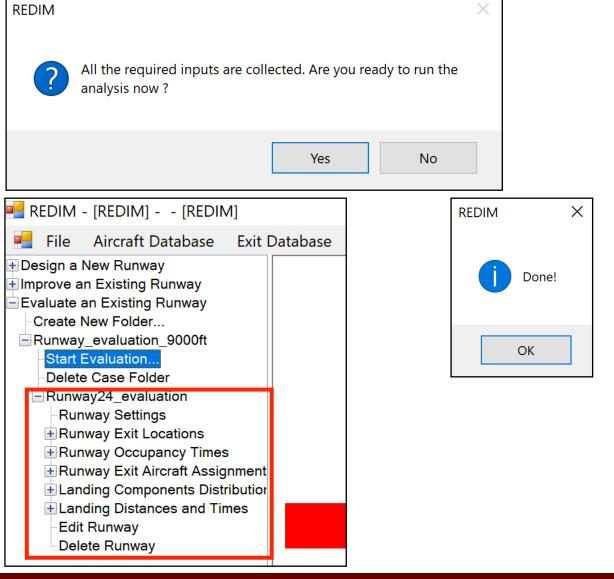
Step 7: Define runway exits



Step by Step Instructions (4)

Step 8: Ready to run the case

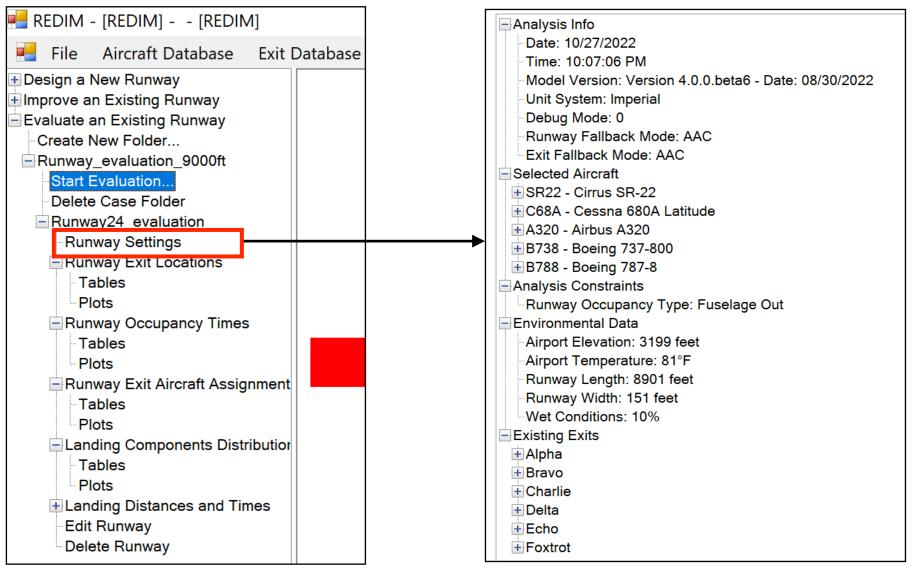
Step 9: Check the outputs



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Step by Step Instructions (5)

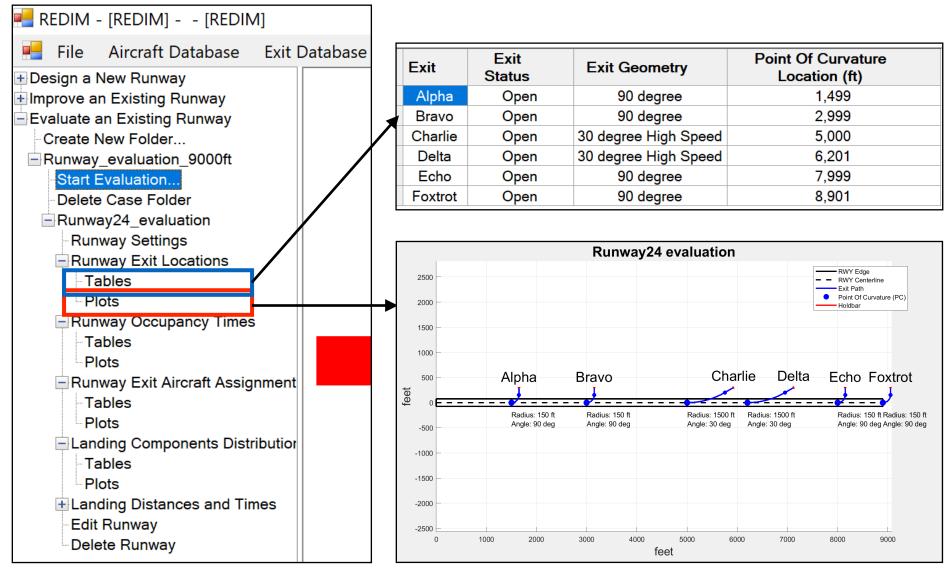


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Step by Step Instructions (6)

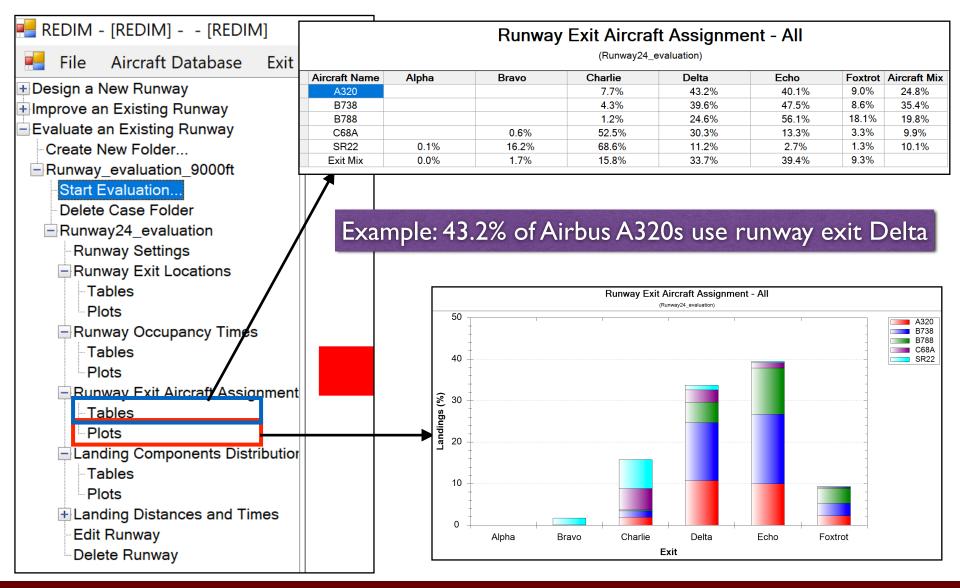


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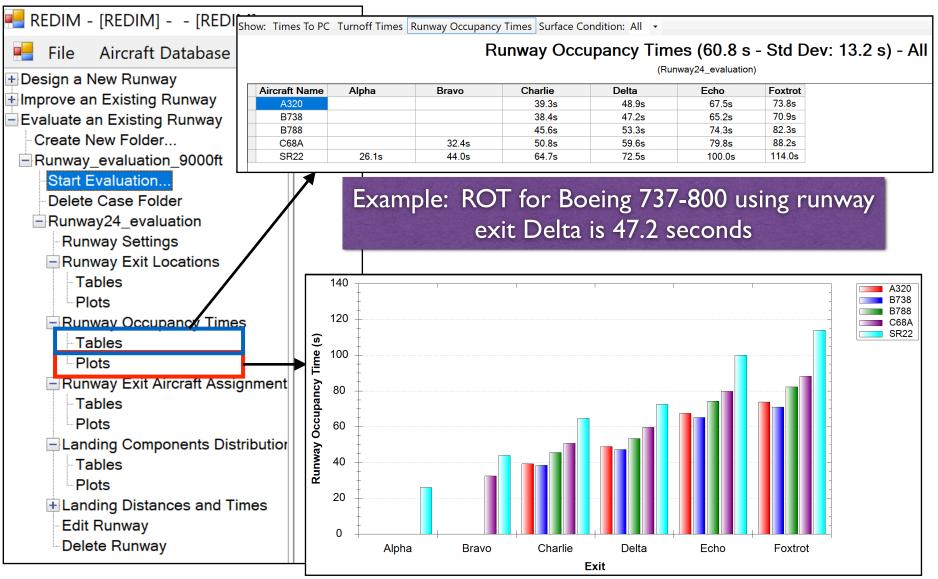


Step by Step Instructions (7)



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Step by Step Instructions (8)









Conclusions

- Airport is located at medium elevation
 - Higher approach speeds (result in longer landing roll out distance)
- Weighted average runway occupancy time is 60.2 seconds
 - Standard deviation of ROT is 13.2 seconds (high)
- High-speed runway exit "Charlie" is underused (16% of time)
- If this was a design exercise, we could move Charlie to 6,000 feet and Delta to 7,000 feet
 - Both acute angle exits could handle 42% and 33% of the traffic, respectively
 - Reduces ROT by a couple of seconds

Another Example Runway Exit Improvements for PHL Airport Using the REDIM 4.0 Model



Application of the Runway Exit Design Tool to PHL

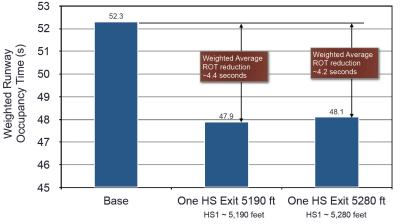
Runway Exit Study High-Speed Runway Exits at PHL Runway 27L	Scenario	Location of New Optimal High- Speed Exit (ft)	Wet/Dry Mix (%/%)	Remarks
	Baseline	Not applicable	10/90	Open exits: U, S7, Y, S9, S11, S12 and S13
N. Mirmohammadsadeghi, N. Hinze and A. Trani November 7, 2019	One High-Speed Runway Exit, 10/90	5,190	10/90	Open exits: U, S7, Y, NewHS1, S11, S12 and S13
	One High-Speed Runway Exit, 20/80	5,280	20/80	Open exits: U, S7, Y, NewHS1, S11, S12 and S13
Air Transportation Systems Laboratory				



- Runway exit Sierra-9 is eliminated
- **793 feet** distance between new exit high-speed exit HS2 and Sierra-11
 Optimally located runway exit in yellow



An Optimally Located High-Speed Runway Exit at PHL Runway 27L Could Reduce the Weighted Average Runway Occupancy Time by 4.4 to 4.2 Seconds

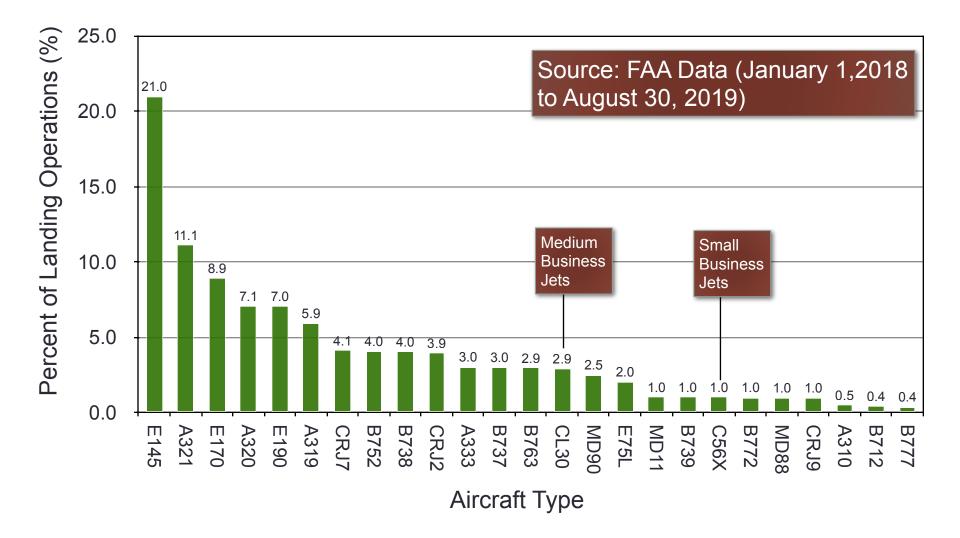




Airport Fleet Mix Used in the Analysis of Runway 27L

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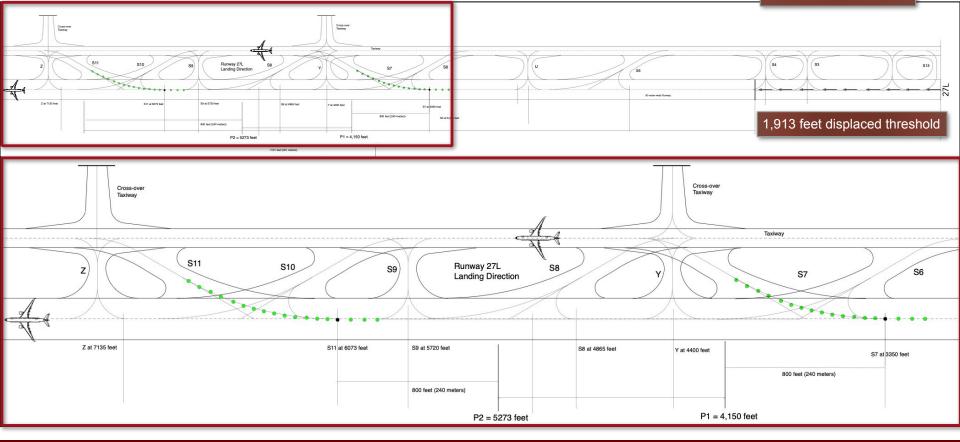


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PHL Runway 27L

- S7 located at 3,350 feet from threshold
- S11 located at 6,073 feet from threshold
- Earliest PC of new high-speed runway exit ~ 4150 feet
- Furthest PC of new high-speed runway exit ~ 5273 feet

If 800 feet is the minimum distance to locate two high-speed exits







Scenarios Studied with New PHL Fleet Mix

Scenario	Location of New Optimal High- Speed Exit (ft)	Wet/Dry Mix (%/%)	Remarks
Baseline	Not applicable	10/90	<i>Open exits: U, S7, Y, S9, S11, S12 and S13</i>
One High-Speed Runway Exit, 10/90	5,190	10/90	<i>Open exits: U, S7, Y, NewHS1, S11, S12 and S13</i>
One High-Speed Runway Exit, 20/80	5,280	20/80	<i>Open exits: U, S7, Y, NewHS1, S11, S12 and S13</i>

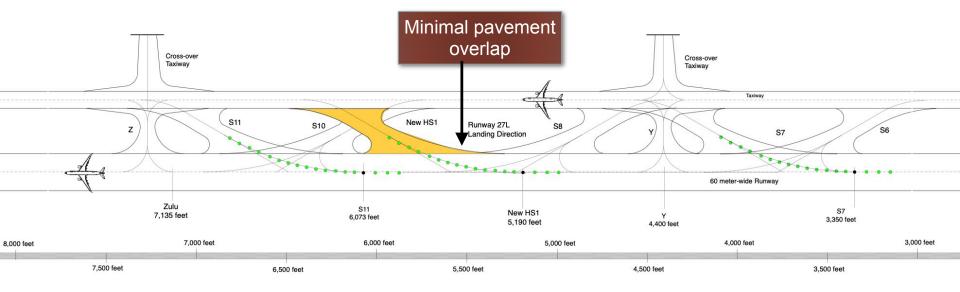
Optimal locations found using REDIM 3 dynamic programming algorithm The backup slides contain probabilities of precipitation at PHL





Case: One New High-Speed Runway Exit, 10/90 (wet/dry pavement design)

- Optimal location of a new High-Speed Runway exit designed for 10/90% wet/dry pavement conditions is <u>5,190</u> <u>feet</u> (point of curvature)
- Runway exit Sierra-9 is eliminated
- 883 feet distance between new exit high-speed exit HS1
 and Sierra-11
 Optimally located runway exit in yellow

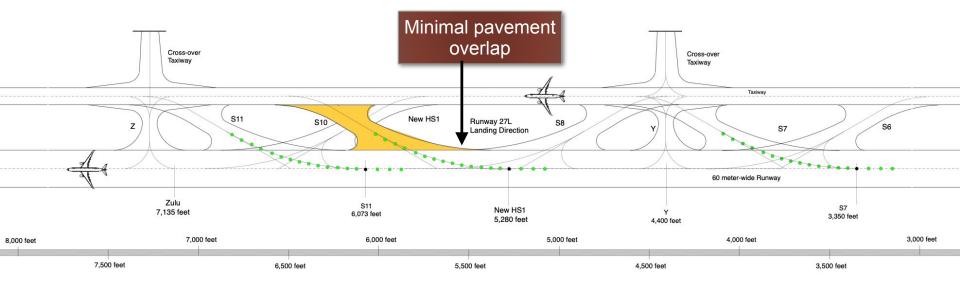




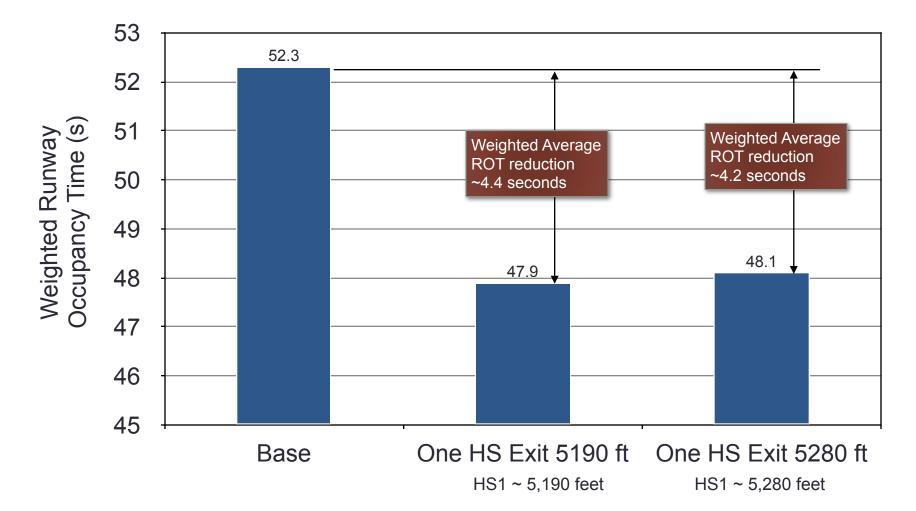


Case: One New High-Speed Runway Exit, 20/80 (wet/dry pavement design)

- Optimal location of a new High-Speed Runway exit designed for 20/80% wet/dry pavement conditions is <u>5,280</u> <u>feet</u> (point of curvature)
- Runway exit Sierra-9 is eliminated
- **793 feet** distance between new exit high-speed exit HS2 and Sierra-11 Optimally located runway exit in yellow



An Optimally Located High-Speed Runway Exit at PHL Runway 27L Could Reduce the Weighted Average Runway Occupancy Time by 4.4 to 4.2 Seconds



PHL Fleet Mix (Jan/2018 to Aug/2019) provided by FAA

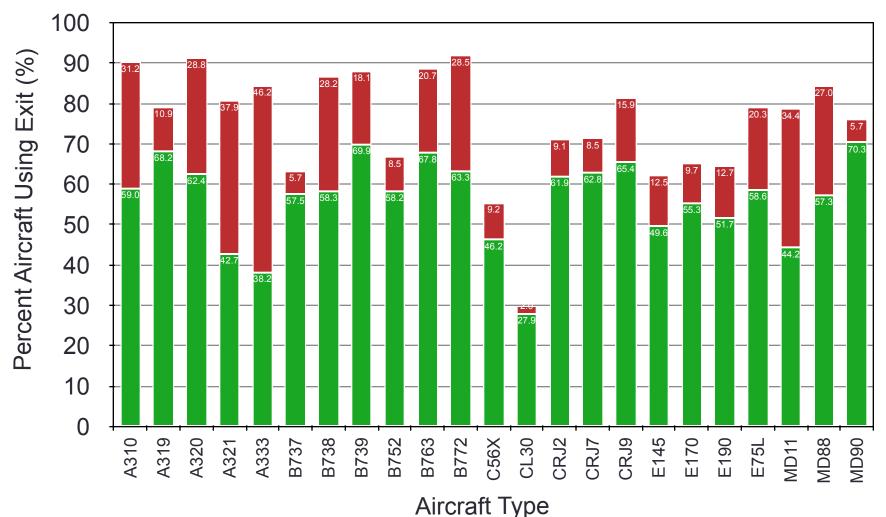
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54% of Landings on Runway 27L Could Use the New High-Speed Exit at 5,280 feet (20/80 wet/dry Design Scenario)

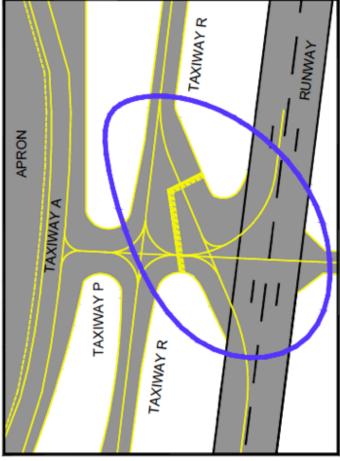
New HS Exit (5280 feet) High-Speed Exit Sierra 11 (6073 feet)



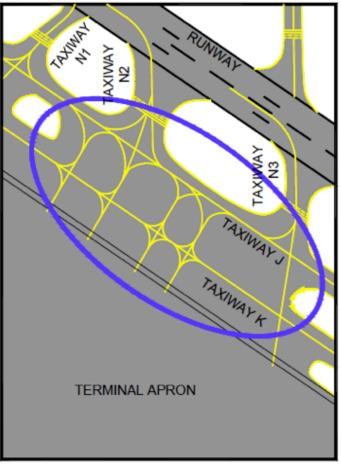
Air Transportation Systems Laboratory

Other Runway Exit and Taxiway Considerations

Things to Avoid (1)



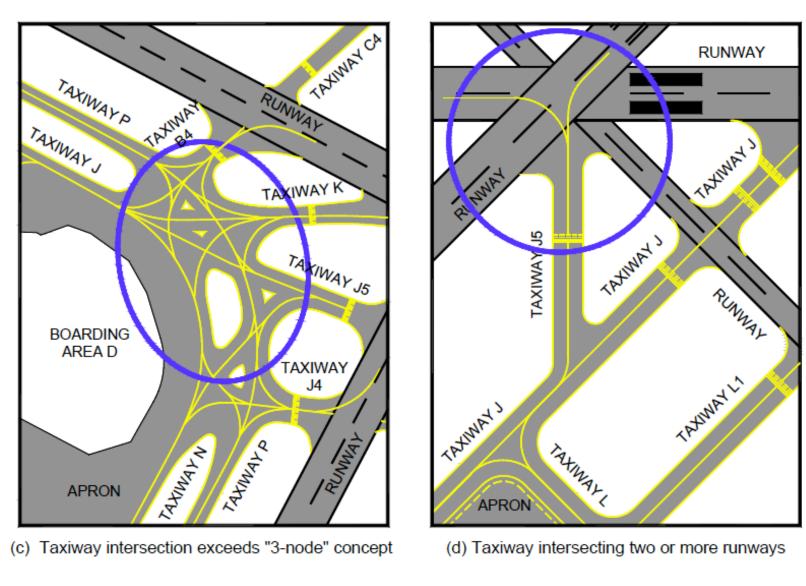
(a) Taxiway crossing high-speed exit and Wide throated runway entrance



(b) Extra-wide throated taxiway without "No Taxi" islands leading from the apron directly to parallel taxiways and runways

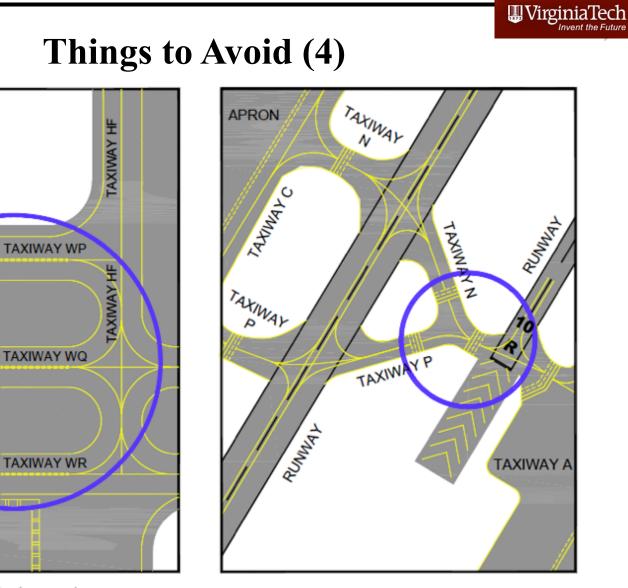
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Things to Avoid (2)



UirginiaTech Things to Avoid (3) TAXIWAY C-2 TAXIWAY C-3 TAXIWAY C-1 > RUNWAY **TAXIWAY D-2** TAXIWAY D RUNWAY TAXIWAY D TAXIWAY D TAXIMAY H

(e) Aligned taxiway between two closely spaced runway ends



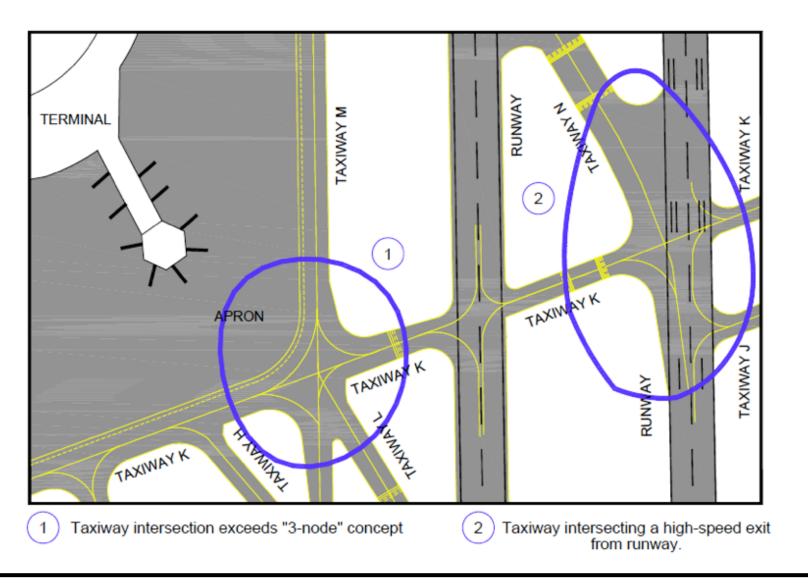
(f) Two or more taxiway entrances lacking "No Taxi" islands

RUNWAY

6

(g) "Y" Shaped taxiway crossing a runway

Things to Avoid (5)



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